

# BeiDou and Galileo, Two Global Satellite Navigation Systems in Final Phase of the Construction, Visibility and Geometry

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**ABSTRACT:** Spatial segment is one of three segments of each satellite navigation systems (SNS). Nowadays two SNSs, GPS and GLONASS, are fully operational, two next SNSs, BeiDou in China and Galileo in Europe, are in final phase of the construction. In the case of China system this segment will consist of 35 satellites with three types of orbits – medium (MEO), geostationary (GEO) and inclined geosynchronous (IGSO). As GEO and IGSO satellites can be used in China and Asia-Pacific region only, BeiDou MEO constellation with 27 fully operational satellites will be taken into account in this paper. The orbital planes of the Galileo constellation will be divided in “slots” that contains at least one operational satellite. The Galileo reference constellation has 24 nominal orbital positions or operational slots in MEO homogeneously distributed in 3 orbital planes; i.e. 8 slots equally spaced per plane. As the error of user’s position obtained from both systems depends on geometry factor DOP (Dilution Of Precision) among other things the knowledge of the number of satellites visible by the user above given masking elevation angle  $H_{min}$  and the distributions of DOP coefficient values, GDOP in particular, is very important. The lowest and the greatest number of satellites visible in open area by the observer at different latitudes for different  $H_{min}$ , the percentage of satellites visible above angle  $H$ , distributions (in per cent) of satellites azimuths and GDOP coefficient values for different  $H_{min}$  for BeiDou and Galileo systems at different latitudes are presented in the paper.

## 1 INTRODUCTION

In early 1980s, when the first satellite navigation system (SNS), Transit in USA and Cikada in Soviet Union were already operational, China began actively study the satellite systems in line with Chinas’s conditions. This new system, called BeiDou (Big Dippler), earlier Compass, as all other SNSs, consists of three major components: the space constellation, the ground control segment and the user segment. In 2000, the BeiDou Navigation Satellite Demonstration System (BDS) was established. (Report...2013, [www.insidegnss.com](http://www.insidegnss.com)).

The functions and performance parameters at this step are as follows (Report...2013):

- position accuracy: better than 10 m (95%);
- velocity accuracy: better than 0.2 m/s;
- time accuracy: 50 ns;
- short message communications: 120 Chinese characters per message.

Global services BeiDou will provide by around 2019/2020 (Munich 2016, [imo.org](http://imo.org)). The final space constellation of this system will consist of 35 satellites:

- 5 GEO (058.75° E, 080° E, 110.5° E, 140° E and 160° E), altitude 35,786 km;

- 27 MEO (altitude 21,500 km, inclination 55°, 3 evenly distributed orbital planes, on each 9 evenly distributed satellites). The phase is selected from the Walker 24/3/1 constellation, and the right ascension of ascending node of the satellites in the first orbital plane is 0°;
- 3 IGSO (altitude 36,000 km, inclination 55°, evenly distributed in 3 orbital planes). The tracks of sub-satellite points for those IGSO satellites are coincided while the longitude of the intersection point is 118° E, with a phase difference of 120°. The sub-satellite tracks for the current (August 2016) other two operational IGSO satellites are coincided while the longitude of the intersection point is at 095° E. These two satellites will not be in final constellation.

It means that the final number of operational satellites (35) will be the greatest among all currently operational (GPS – 32, GLONASS – 24) or under construction (Galileo – 30) SNSs. BeiDou constellation, as the only SNS, uses three mentioned above types of orbits, while all others use MEO orbits only (Januszewski J. 2010, Martin H. 2013).

The most important parameters of all BeiDou satellites are presented in the table 1.

Table 1. BeiDou system, parameters of GEO, IGSO and MEO satellites (www.isg.org)

Parameter	Satellite		
	GEO	IGSO	MEO
Satellite bus	DFH-3B	DFH-3B	Navigation satellite bus
Launch vehicle	CZ-3C	CZ-3A	CZ-3B
Launch mass [kg]	4600	4200	800
Dry mass [kg]	1550	1900	no details
Body size [m]	1.8 x 2.2 x 2.5		
Solar array size [m]	3 x 2.2 x 1.7 (two pieces)		
Span width [m]	17.7	17.7	17.7
Cross section [m <sup>2</sup> ]	27	27	27
Power output [kW]	6.8	6.2	1.5
Space design line [years]	8	8	5

The satellites are and will be launched from the XSLC (Xichang Satellite Launch Center) in the southwestern province Sichuan of China, using China's Long March 3 CZ launch vehicles (www.beidou.gov.cn).

The BeiDou navigation (NAV) messages are formatted in D1 and D2 based on their rate and structure in superframe, frame and subframe. The first format will be transmitted by all 35 satellites, the second is already by GEO satellites only. The rate of these messages is 50 bps and 500 bps respectively (www.beidou.gov.cn).

D1 NAV message contains basic, named sometimes fundamental also, information about broadcasting 27 MEO and 3 IGSO satellites, almanac information offsets from other global SNSs (GPS, GLONASS and Galileo).

D2 NAV message contains mentioned above basic NAV information of the broadcasting satellite and additionally augmentation service information – integrity and differential correction information, ionospheric grid information. In the case of this format the definition of basic NAV information is the

same as that in format D1 except the page number (Pnum) and seconds of week (SOW), which are different from these in format D1 (BeiDou, Signal. 2013, BeiDou, Open. 2013).

In Europe the development of initial design concepts for a new own satellite navigation system began in the 1990s, with selection of an initial system design in 2002. In March 2002 the European Union and the European Space Agency agreed to fund the project, named Galileo, pending a review in 2003. The Galileo constellation design was originally planned based on a three-plane Walker constellation with a minimum of nine operational satellites in each plane and three active spares, one per orbital plane; total number of satellites – 30.

Over the course of time the planned Galileo constellation changed and in 2014 these changes became evident. The reference space segment has 24 operational satellites only in a Walker 24/3/1 design along with up to six operating spares – two in each plane. The eight satellites in each plane are equally spaced; locations of the spares will be determined. The MEO planes, altitude 23,222 km, have 120° separation with inclination 56°. Galileo satellites nominally have a mass of 700 kg, measuring 2.7 m x 1.1 m x 1.2 m, with deployed solar arrays spanning 18.7 m (Beitz J.W. 2016, www.gsc.europa.eu).

All Galileo satellites were launched, two satellites at once, from the Guiana Space Centre, French and European spaceport near Kourou in French Guiana, using carrier Soyuz-STB/Fregat-MT. Currently (August 2016) spatial segment consists of 9 satellites, one partially unavailable, four with status in commissioning (two last satellites launched May 24, 2016 will be operational soon). Next launch is scheduled for November 17, 2016 from Kourou also but using new carrier Ariane 5ES. This dispenser allows for a launch of four satellites at once. The subsequent eight satellites will be launched in 2017 and 2018, the last one in 2019.

## 2 BEIDOU AND GALILEO CONSTELLATIONS

The nominal values of geographical longitude of ascending node and argument of latitude of all 27 BeiDou MEO satellites and all 24 Galileo are showed in the table 2 and table 3, respectively.

Table 2. BeiDou system, geographical longitude of ascending node (An) and argument of latitude (Al) of all 27 MEO satellites

Orbit I, An = 0°		Orbit II, An = 120°		Orbit III, An = 240°	
No of satellite	Al [ deg ]	No of satellite	Al [ deg ]	No of satellite	Al [ deg ]
1	0	10	13	19	26
2	40	11	53	20	66
3	80	12	93	21	106
4	120	13	133	22	146
5	160	14	173	23	186
6	200	15	213	24	226
7	240	16	253	25	266
8	280	17	293	26	306
9	320	18	333	27	346

Table 3. Galileo system, geographical longitude of ascending node (An) and argument of latitude (Al) of all 24 satellites

Orbit I, An = 0°		Orbit II, An = 120°		Orbit III, An = 240°	
No of satellite	Al [ deg ]	No of satellite	Al [ deg ]	No of satellite	Al [ deg ]
1	0	9	15	17	30
2	45	10	60	18	75
3	90	11	105	19	120
4	135	12	150	20	165
5	180	13	195	21	210
6	225	14	240	22	255
7	270	15	285	23	300
8	315	16	330	24	345

### 3 TEST METHODS

All calculations based on reference ellipsoid WGS-84 were made on author's simulating program. The interval of the latitude of the observer between 0° and 90° was divided into 9 zones, each 10° wide. In the observer's receiver masking elevation angle  $H_{min}$  was assumed to be 0° (horizon), 5° (the most frequently used value of  $H_{min}$ ), 10°, 15°, 20° and 25°. The angle 25° is representative for the positioning in restricted area where the visibility of satellites can be limited. This problem is very important in road transport (urban canyon) and in maritime transport in restricted area where the visibility of satellites is limited. The calculations were made for 27 BeiDou MEO and 24 Galileo satellites constellations.

For each zone of latitude and for each masking elevation angle ( $H_{min}$ ) one thousand (1000) geographic-time coordinates of the observer were generated by random-number generator with uniform distribution:

- latitude interval 0 – 600 minutes (10°),
- longitude interval 0 – 21600 minutes (360°),
- time interval in minutes, Galileo: 0 – 14,360.75, (17 orbital periods, each 14 h 4 min 45 sec), BeiDou: 0 – 10,091.48 minutes (13 MEO satellite orbital periods, each 12 h, 56 min 16.05 sec).

For each geographic-time coordinates: the satellite elevation (H), the satellite azimuth (Az), the number of visible satellites (Is) and GDOP (Geometric Dilution of Precision) coefficient values were calculated. Elevation H was divided into 9 intervals, each 10° wide, azimuth (Az) was divided into 8 intervals, each 45° and GDOP value (w) into 8 intervals:  $w < 2$ ,  $2 \leq w < 3$ ,  $3 \leq w < 4$ ,  $4 \leq w < 5$ ,  $5 \leq w < 6$ ,  $6 \leq w < 8$ ,  $8 \leq w < 20$ ,  $w \geq 20$ .

### 4 VISIBILITY OF BEIDOU MEO SATELLITES AND GALILEO SATELLITES

The lowest ( $Is_{min}$ ) and the greatest ( $Is_{max}$ ) number Is of satellites visible by the observer in open area above different  $H_{min}$  in all 9 zones of latitudes for both systems are presented in the table 4. As for  $H_{min} \leq 25^\circ$  the number Is is in the case of BeiDou greater than 3 for all user's latitudes it means that 3D position can be obtained always and anywhere in the world. In the case of Galileo the number Is can be equal 3 in

zone 0–10° and at latitudes 40–60°; it means 2D position only. We can say also that:

- for both systems the number Is depends on observer's latitude, independently of  $H_{min}$  value;
- for  $H_{min} = 0^\circ$  the number  $Is_{min}$  is for BeiDou system the lowest (6) in zone 20–30° and the greatest (9) in zone 0–10° and at latitudes 60–90°, for Galileo system this number is the lowest (6 also) at latitudes 20–60° and the greatest (9 also) at latitudes 70–90°; the number  $Is_{max}$  (12 always) is for BeiDou in all 9 latitude zones, for Galileo at latitudes 10–40° and 60–90°.

If the number Is is equal 4 the user must be very careful because the number of satellites which can be used in position determination can decrease at any moment and for any reason. That's why the additional calculations were made for both systems for different  $H_{min}$  in order to determine the greatest elevation H for which the number Is of satellites visible at different latitudes in open area above this angle is equal 4 (Is4) or 3 (Is3) (table 5).

Table 4. System BeiDou, the lowest and the greatest number of MEO satellites visible in open area above  $H_{min}$  at different observer's latitudes.

Latitude [°]		$H_{min}$					
		0°	5°	10°	15°	20°	25°
0–10	BeiDou	9–12	8–12	7–10	5–10	4–8	4–7
	Galileo	8–11	8–11	6–10	4–8	4–8	3–7
10–20	BeiDou	8–12	6–12	6–10	5–9	4–9	4–8
	Galileo	7–12	6–11	5–9	4–9	4–8	4–8
20–30	BeiDou	6–12	6–11	6–10	5–9	4–9	4–7
	Galileo	6–10	6–10	5–9	4–9	4–8	4–7
30–40	BeiDou	7–12	6–12	6–10	5–9	5–9	4–7
	Galileo	6–12	6–11	5–9	4–9	4–9	4–7
40–50	BeiDou	7–12	6–12	6–10	5–10	4–8	4–7
	Galileo	6–11	6–11	5–9	4–9	4–8	3–7
50–60	BeiDou	8–12	6–11	6–10	5–10	4–9	4–7
	Galileo	6–10	6–10	5–9	4–9	4–7	3–7
60–70	BeiDou	9–12	8–12	7–11	6–9	5–9	4–7
	Galileo	8–12	8–11	6–9	5–9	4–8	4–7
70–80	BeiDou	9–12	8–12	8–11	6–9	5–8	4–8
	Galileo	9–12	8–11	7–10	5–9	5–8	4–7
80–90	BeiDou	9–12	9–12	8–10	7–9	6–9	4–8
	Galileo	9–12	8–11	7–9	6–8	6–8	4–7

Table 5. BeiDou system (MEO satellites only) and Galileo system, the greatest elevation H [°] for which the number Is of satellites visible at different latitudes by the observer in open area above this angle is equal 4 or 3

Latitude [°]	Number Is of satellites			
	4		3	
	BeiDou	Galileo	BeiDou	Galileo
0 – 10	26	23	32	29
10 – 20	28	25	33	29
20 – 30	30	26	35	33
30 – 40	28	26	33	30
40 – 50	28	24	33	32
50 – 60	28	24	32	33
60 – 70	28	25	32	32
70 – 80	29	27	32	29
80 – 90	33	32	34	35

If the number Is is equal 4 the user must be very careful because the number of satellites which can be used in position determination can decrease at any moment and for any reason. That's why the additional calculations were made for both systems

for different  $H_{\min}$  in order to determine the greatest elevation  $H$  for which the number  $l_s$  of satellites visible at different latitudes in open area above this angle is equal 4 ( $l_{s4}$ ) or 3 ( $l_{s3}$ ) (table 5).

In the case of BeiDou system and  $l_{s4}$  the elevation  $H$  is the lowest (26) in zone  $0-10^\circ$ , the greatest (33) in zone  $80-90^\circ$ . In the case of  $l_{s3}$  the lowest elevation  $H$  (32) is in zone  $0-10^\circ$  and at latitudes  $50-80^\circ$ , the greatest (34) in zone  $20-30^\circ$ . In the case of Galileo and  $l_{s4}$  the  $H$  is the lowest (23) in zone  $0-10^\circ$ , the greatest (32) in zone  $80-90^\circ$ , and for  $l_{s3}$  the lowest (29) at latitudes  $0-20^\circ$  and in zone  $70-80^\circ$ , the greatest (35) in zone  $80-90^\circ$ .

The weighted mean number  $l_{sm}$  of BeiDou MEO satellites and Galileo satellites visible above horizon and the percentage of satellites visible by the observer in open area above given angle  $H$  in all 9 latitude zones for both systems are showed in the table 6. Supplementary calculations were made for  $H = 5^\circ, 15^\circ$  and  $25^\circ$  in all these zones. Because of greater number of satellites (BeiDou – 27, Galileo – 24) the  $l_{sm}$  is for China system greater than for European system irrespective of latitude.

Additionally we can say that for both systems:

- the percentage of satellites visible decreases with angle  $H$  in each zone, if this angle is equal  $20^\circ$  this percentage decreases to about 60% or more, if it is  $40^\circ$  to about 30% or more;
- at latitudes equal or greater than  $70^\circ$  the percentage of satellites visible above  $H \leq 20^\circ$  is greater than in all other latitudes;
- at latitudes less than  $70^\circ$  the percentage of satellites visible above  $20^\circ \leq H \leq 50^\circ$  is the greatest in zone  $30-40^\circ$ ;
- because of altitude and orbit inclination values of both systems and geometrical figure of the Earth in zone  $70-80^\circ$  any BeiDou and Galileo satellite cannot be visible by the observer above  $70^\circ$  and

$80^\circ$  respectively, in zone  $80-90^\circ$  any satellite of both systems above  $60^\circ$ .

Distribution (in per cent) of satellites azimuths in open area for  $H_{\min} = 5^\circ$  and  $25^\circ$  in all 9 latitudes zones of hemisphere north for both systems is presented in the table 7. We can resume that for both systems:

- for each  $H_{\min}$  and each zone both distributions are almost the same;
- distribution of satellites azimuths depends on observer's latitude and angle  $H_{\min}$ ;
- at latitudes  $0-20^\circ$  the percentage of satellites with azimuths from intervals  $045-135^\circ$  and  $225-315^\circ$  is for  $H_{\min} = 5^\circ$  less than in all other intervals;
- at latitudes  $30-50^\circ$  the percentage of satellites in interval  $45-90^\circ$  and interval  $270-315^\circ$  is for both  $H_{\min}$  values greater than in all other intervals considerably;
- at latitudes  $50-70^\circ$  in all 6 intervals of azimuths  $45-315^\circ$  the percentage of satellites increases with  $H_{\min}$  whereas in two other intervals (azimuth  $315-045^\circ$ ) decreases significantly;
- at latitudes  $40-70^\circ$  the percentage of satellites in intervals  $45-90^\circ$  and  $270-315^\circ$  is for  $H_{\min} = 25^\circ$  several dozen times greater than in interval  $315-045^\circ$ ;
- at latitudes  $80-90^\circ$  the percentage of satellites in all intervals is for both  $H_{\min}$  almost the same.
- the percentage of satellites is almost the same in all 9 zones however for Galileo system for each  $H \leq 60^\circ$  it is greater than for BeiDou system;
- at latitudes  $70-80^\circ$  the number of satellites is in all intervals almost the same, but for  $H_{\min} = 5^\circ$  only.

The percentage of satellites azimuths in different intervals depends on observer's latitude. The greatest diversifications are at latitudes  $30-70^\circ$  for both systems. That's why the knowledge of these distributions are very important for the users, in road transport in urban canyon in particular.

Table 6. Percentage of BeiDou MEO satellites and Galileo satellites visible in open area above angle ( $H$ ) at different observer's latitudes ( $\varphi$ ),  $l_{sm}$  – weighted mean number of satellites visible above horizon ( $H = 0^\circ$ )

Latitude [°]	System	$l_{sm}$	Elevation angle $H$ [°]								
			0	10	20	30	40	50	60	70	80
0 – 10	BeiDou	10.9	100	81.0	59.7	39.7	25.8	15.5	8.2	3.7	0.9
	Galileo	9.8	100	80.8	59.4	39.7	25.8	15.7	8.0	3.4	0.9
10 – 20	BeiDou	10.7	100	77.0	58.7	43.7	29.0	16.9	9.0	4.0	1.0
	Galileo	9.6	100	76.9	58.4	43.4	29.5	17.0	8.9	3.9	1.0
20 – 30	BeiDou	10.0	100	78.1	61.5	46.9	34.4	21.6	11.4	3.7	1.3
	Galileo	9.1	100	78.3	61.2	46.9	34.3	21.7	11.1	4.9	1.2
30 – 40	BeiDou	9.7	100	79.1	62.3	48.0	35.6	25.0	15.2	6.2	1.5
	Galileo	8.8	100	79.5	63.1	48.3	35.4	25.0	15.1	6.3	1.6
40 – 50	BeiDou	9.9	100	77.7	60.1	45.9	34.2	24.5	15.7	8.2	2.3
	Galileo	9.0	100	77.4	60.7	46.5	34.3	24.0	15.9	8.7	2.1
50 – 60	BeiDou	10.6	100	75.9	56.4	42.3	30.7	21.2	13.1	6.7	2.0
	Galileo	9.6	100	76.6	56.9	43.3	30.7	21.2	13.4	7.2	2.6
60 – 70	BeiDou	10.9	100	77.6	60.0	40.5	27.6	17.6	9.2	3.0	0.2
	Galileo	9.8	100	81.6	57.4	42.0	28.7	18.4	9.7	3.9	0.4
70 – 80	BeiDou	11.0	100	82.9	64.5	42.5	27.7	12.4	3.1	0	0
	Galileo	10.0	100	83.2	66.0	45.3	26.7	13.6	4.3	0.2	0
80 – 90	BeiDou	11.0	100	83.5	66.6	48.6	25.1	3.4	0	0	0
	Galileo	10.0	100	83.7	67.6	50.4	29.2	5.6	0	0	0

Table 7. Distribution (in per cent) of BeiDou MEO satellites and Galileo satellites azimuths in open area for different masking elevation angles ( $H_{\min}$ ) at different observer's latitudes ( $\varphi$ ),  $l_m$  – weighted mean number of satellites visible above  $H_{\min}$

$\varphi$ [ $^{\circ}$ ]	N	System	$H_{\min}$ [ $^{\circ}$ ]	$l_m$	Azimuth [ $^{\circ}$ ]						
					0 – 45	45 – 90	90 – 135	135 – 180	180 – 225	225 – 270	270 – 315
0–10	BeiDou	5	9.9	15.2	10.0	10.0	15.0	14.7	10.1	10.3	14.7
		25	5.4	17.3	10.1	10.4	12.7	11.7	10.0	10.5	17.3
	Galileo	5	8.9	14.5	10.3	9.8	15.1	15.3	9.8	10.3	14.9
		25	4.8	16.5	11.0	10.0	12.3	12.2	9.7	10.8	17.5
10–20	BeiDou	5	9.5	14.8	11.2	10.2	14.1	13.6	10.3	11.5	14.3
		25	5.5	18.7	10.5	10.5	10.7	10.4	9.9	11.0	18.3
	Galileo	5	8.6	14.6	11.3	9.8	14.0	14.0	9.9	11.4	15.0
		25	4.9	18.2	10.9	9.9	10.5	10.5	11.7	11.7	18.6
20–30	BeiDou	5	8.9	13.3	13.7	10.8	12.2	11.8	10.6	13.9	13.7
		25	5.4	16.9	12.7	10.5	9.8	10.2	10.0	13.3	16.6
	Galileo	5	8.0	14.1	13.2	10.6	11.8	11.8	10.6	13.8	14.1
		25	4.9	17.0	12.3	10.2	10.2	10.3	10.0	13.2	16.8
30–40	BeiDou	5	8.7	9.6	18.0	11.1	11.4	11.2	11.0	18.1	9.8
		25	5.3	10.4	18.2	11.0	10.0	10.6	10.6	18.7	10.5
	Galileo	5	7.8	10.6	17.2	10.9	11.1	11.4	10.9	16.8	11.1
		25	4.9	11.5	17.4	10.9	9.7	10.5	10.6	17.3	12.1
40–50	BeiDou	5	8.7	5.7	21.2	11.6	11.4	11.2	11.9	21.4	5.6
		25	5.2	2.9	23.5	11.9	11.6	10.7	12.8	23.7	2.9
	Galileo	5	7.9	6.6	20.8	11.4	10.9	11.3	11.6	20.4	7.0
		25	4.8	3.8	23.5	11.8	10.5	11.0	12.3	23.0	4.1
50–60	BeiDou	5	9.3	7.4	18.3	12.8	11.5	11.3	13.1	17.8	7.8
		25	5.1	0.3	21.4	14.9	13.3	12.6	16.3	20.8	0.4
	Galileo	5	8.5	8.7	18.3	12.3	10.8	11.2	12.3	17.8	8.6
		25	4.8	0.8	22.3	14.1	12.0	12.7	15.4	21.7	1.0
60–70	BeiDou	5	9.8	10.1	14.3	13.7	12.0	12.0	13.3	14.2	10.4
		25	5.3	1.5	16.0	17.5	15.0	14.8	17.7	16.1	1.4
	Galileo	5	9.0	10.8	14.6	12.8	11.8	12.2	12.8	14.3	10.7
		25	4.9	2.0	16.8	16.1	14.6	15.1	16.4	16.6	2.4
70–80	BeiDou	5	10.0	11.3	13.1	12.9	12.7	12.4	12.7	13.1	11.8
		25	6.0	7.7	13.2	14.6	14.4	14.4	14.3	13.5	7.9
	Galileo	5	9.1	11.8	13.1	12.7	12.2	12.7	12.5	13.1	11.9
		25	5.6	8.9	13.5	14.2	13.3	14.0	14.0	13.3	8.8
80–90	BeiDou	5	10.1	12.1	12.8	12.6	12.4	12.8	12.1	13.0	12.2
		25	6.4	11.4	12.6	12.9	13.0	13.4	12.2	13.0	11.5
	Galileo	5	9.2	12.3	12.5	12.5	12.4	13.0	12.4	12.5	12.4
		25	5.9	12.5	11.8	13.3	12.8	13.0	13.1	11.8	11.7

## 5 GEOMETRY OF BEIDOU MEO SATELLITES AND GALILEO SATELLITES

Distribution (in per cent) of GDOP coefficient values in open area for  $H_{\min} = 5^{\circ}$  and  $25^{\circ}$  in all 9 latitudes zones for both systems is presented in the table 8. Additional calculations were made in zone  $50-60^{\circ}$  (latitude of Poland) for four other  $H_{\min}$  values ( $0^{\circ}$ ,  $10^{\circ}$ ,  $15^{\circ}$  and  $20^{\circ}$ ). We can say that GDOP coefficient value:

- depends on angle  $H_{\min}$  and observer's latitudes and its distributions are almost the same for both systems;
- increases with  $H_{\min}$  in all 9 zones for both systems, but the greatest values and growths are at high latitudes;
- can be less than 2 for BeiDou system and  $H_{\min} = 5^{\circ}$  at latitudes  $0-20^{\circ}$  only;
- for  $H_{\min} = 5^{\circ}$  is always less than 4 for BeiDou in zone  $0-10^{\circ}$  and for Galileo in zone  $20-40^{\circ}$ ;
- for  $H_{\min} = 5^{\circ}$  for both systems is always greater than 3 in zone  $70-80^{\circ}$  and greater than 4 in zone  $80-90^{\circ}$ ;
- for  $H_{\min} = 25^{\circ}$  can be less than 3 for Galileo in zone  $10-20^{\circ}$  only and for BeiDou in zone  $60-70^{\circ}$  only;

- for  $H_{\min} = 25^{\circ}$  can be equal or greater than 20 in all 9 zones for both systems;
- for  $H_{\min} = 25^{\circ}$  and for both systems is equal at least 4 or greater in zone  $70-80^{\circ}$  and equal 6 or greater in zone  $80-90^{\circ}$ .

As was showed in the table 4 in the case of Galileo in three latitude zones,  $0-10^{\circ}$ ,  $40-50^{\circ}$  and  $50-60^{\circ}$ , if  $H_{\min} = 25^{\circ}$  3D position cannot be obtained always. When the number of satellites visible by the observer decreases to three the number of fix without position (No fix) is greater than zero.

Distribution (in per cent) of GDOP coefficient values for the observer at latitudes  $50-60^{\circ}$ , if the number of visible satellites  $l_s$  is known, for  $H_{\min} = 25^{\circ}$  and  $5^{\circ}$ , for both systems is presented respectively in the table 9 and the table 10. We can showed that:

- for BeiDou system if  $H_{\min} = 5^{\circ}$  GDOP value is less than 3 if  $l_s = 11$  and can be less than 3 if  $l_s = 8, 9$  or 10. If  $l_s = 6$  this coefficient is equal or greater than 3 and less than 5;
- for Galileo system if  $H_{\min} = 5^{\circ}$  GDOP value is less than 3 if  $l_s = 10$ , can be less than 3 if  $l_s = 7, 8$  or 9. If  $l_s = 6$  this coefficient is equal or greater than 3 and less than 4;

- for BeiDou system if  $H_{\min} = 25^\circ$  GDOP value is less than 4 if  $l_s = 7$  and can be less than 4 if  $l_s = 6$  or 5. If  $l_s = 4$  this coefficient is equal at least 8;
- for Galileo system if  $H_{\min} = 25^\circ$   $l_s$  can be equal 3 (0.3 %) and then the position 3D cannot be obtained. GDOP value is less than 4 if  $l_s = 7$  and can be less than 4 if  $l_s = 5$  or 6. If  $l_s = 4$  this coefficient is equal at least 5.
- there is not a direct relation between a number  $l_s$  of satellites visible above  $H_{\min}$  and GDOP coefficient value, but in the case of both systems we can realize "when  $l_s$  is greater, GDOP can be less" and vice versa "when  $l_s$  is less, GDOP can be greater".

Table 8. BeiDou MEO satellites and Galileo satellites distribution (in per cent) of GDOP coefficient values and No Fix (without position) for different masking elevation angles ( $H_{\min}$ ) at different observer's latitudes ( $\varphi$ )

Latitude [°]	System	$H_{\min}$ [°]	No Fix	GDOP coefficient value - w							
				w<2	2≤w<3	3≤w<4	4≤w<5	5≤w<6	6≤w<8	8≤w<20	w≥20
0-10	BeiDou	5	-	0.6	91.3	8.1	-	-	-	-	-
		25	-	-	-	25.8	37.7	15.0	7.0	9.4	5.1
	Galileo	5	-	-	93.5	5.2	1.3	-	-	-	-
		25	0.9	-	-	17.6	27.5	5.6	10.1	25.5	12.8
10-20	BeiDou	5	-	0.4	86.3	12.4	0.9	-	-	-	-
		25	-	-	-	16.0	41.3	14.7	8.3	16.9	2.8
	Galileo	5	-	-	89.7	9.9	0.4	-	-	-	-
		25	-	-	0.7	8.5	32.7	8.2	11.6	27.5	10.8
20-30	BeiDou	5	-	-	76.1	17.4	6.5	-	-	-	-
		25	-	-	-	18.7	34.4	16.8	8.2	13.5	8.4
	Galileo	5	-	-	69.2	30.8	-	-	-	-	-
		25	-	-	-	2.2	47.0	7.5	11.2	24.7	7.4
30-40	BeiDou	5	-	-	95.8	2.7	1.5	-	-	-	-
		25	-	-	-	11.2	44.7	36.7	1.6	5.6	0.2
	Galileo	5	-	-	54.7	45.3	-	-	-	-	-
		25	-	-	-	10.8	46.1	2.2	5.7	17.5	17.7
40-50	BeiDou	5	-	-	86.4	7.8	5.8	-	-	-	-
		25	-	-	-	24.5	27.2	28.5	-	14.9	4.9
	Galileo	5	-	-	51.5	47.3	1.2	-	-	-	-
		25	0.1	-	-	27.5	18.4	1.6	4.7	18.4	29.3
50-60	BeiDou	5	-	-	70.4	27.8	1.8	-	-	-	-
		25	-	-	-	13.6	39.0	22.4	0.1	14.7	10.2
	Galileo	5	-	-	69.6	30.1	0.3	-	-	-	-
		25	0.3	-	-	17.7	45.0	0.3	2.8	11.1	22.8
60-70	BeiDou	5	-	-	28.1	69.3	2.6	-	-	-	-
		25	-	-	0.1	1.8	29.9	49.2	5.7	8.4	4.9
	Galileo	5	-	-	19.2	79.5	1.3	-	-	-	-
		25	-	-	-	1.9	37.7	15.9	13.5	15.5	15.5
70-80	BeiDou	5	-	-	-	35.7	49.1	14.4	0.8	-	-
		25	-	-	-	-	8.8	27.1	27.0	35.5	1.6
	Galileo	5	-	-	-	31.1	54.4	14.2	0.3	-	-
		25	-	-	-	-	9.4	23.1	24.6	41.5	1.4
80-90	BeiDou	5	-	-	-	-	0.8	13.1	27.2	37.1	21.8
		25	-	-	-	-	-	-	10.8	60.6	28.6
	Galileo	5	-	-	-	-	0.6	11.5	26.6	39.0	22.3
		25	-	-	-	-	-	-	7.2	61.8	31.0

Table 9. BeiDou MEO satellites and Galileo satellites, distribution (in per cent) of GDOP coefficient values at observer's latitudes 50-60°, if the number of visible satellites  $l_s$  is known,  $H_{\min} = 25^\circ$ , No fix - the percentage of fix without position

System and satellites	$l_s$	System	%	GDOP coefficient value - w				
				3≤w<4	4≤w<5	5≤w<6	6≤w<8	8≤w<20
No Fix	BeiDou	-	-	-	-	-	-	-
		Galileo	0.3	-	-	-	-	-
4	BeiDou	24.7	-	-	-	-	14.6	10.1
		Galileo	36.1	-	-	0.2	2.1	11.0
5	BeiDou	39.3	0.1	16.5	22.4	0.1	0.1	0.1
		Galileo	47.9	3.0	44.0	0.1	0.7	0.1
6	BeiDou	32.7	10.2	22.5	-	-	-	-
		Galileo	13.7	12.7	1.0	-	-	-
7	BeiDou	3.3	3.3	-	-	-	-	-
		Galileo	2.0	2.0	-	-	-	-
	BeiDou	100	13.6	39.0	22.4	0.1	14.7	10.2
		Galileo	100	17.7	45.0	0.3	2.8	11.7

Table 10. BeiDou MEO satellites, Galileo satellites, distribution (in per cent) of GDOP coefficient values at observer's latitudes  $50 - 60^\circ$ ,  $H_{\min} = 5^\circ$ , if the number of visible satellites  $l_s$  is known

System	Visible satellites		GDOP coefficient value – w		
	$l_s$	%	$2 \leq w < 3$	$3 \leq w < 4$	$4 \leq w < 5$
BeiDou 6	6	2.1	–	0.3	1.8
Galileo		2.0	–	2.0	–
BeiDou 7	7	3.5	–	3.5	–
Galileo		7.7	1.0	6.4	0.3
BeiDou 8	8	14.9	3.3	11.6	–
Galileo		36.2	15.1	21.1	–
BeiDou 9	9	28.1	16.8	11.3	–
Galileo		44.5	43.8	0.7	–
BeiDou 10	10	45.0	43.9	1.1	–
Galileo		9.6	9.6	–	–
BeiDou 11	11	6.4	6.4	–	–
Galileo		–	–	–	–
		100	70.4	27.8	1.8
		100	69.6	30.1	0.3

As the longitudes of two extreme BeiDou GEO satellites are  $058.75^\circ$  E and  $160^\circ$  E, the trucks of sub-satellite points for 3 IGSO satellites are coincided and the longitude of the intersection points is  $118^\circ$  E, this system currently provides positioning data between longitude  $055^\circ$  E to  $180^\circ$  E and from latitude  $55^\circ$  S to  $55^\circ$  N. This area includes China territory entirely.

## 6 BEIDOU AND GALILEO STATUS

Currently (August 2016) there are 20 BeiDou satellites in orbit and healthy and one with status in commissioning and 9 Galileo operational satellites, 4 with status in commissioning and one partially unavailable. By 2020 or earlier BeiDou would reach full operational capability (FOC) with 35 satellites. In the case of Galileo 12 next satellites are planned to be launched in the years 2016–2018, the last in 2019. FOC of this system is planned in 2020 (European GNSS Service Centre, [www.gpsworld.com](http://www.gpsworld.com), [www.gsc.europa.eu](http://www.gsc.europa.eu)).

## 7 CONCLUSIONS

- Full Operational Capability (FOC) of BeiDou system will made China the third nation in possession of independent, global navigation system following the United States and Russia. FOC of Galileo will made the first in history international institution (European Union) in possession of global system.
- BeiDou MEO constellation with 27 fully operational satellites will provide global coverage and the possibility of user's 3D position determination at any moment and any point on the Earth as well as in the case when the satellites are visible by the user below  $25^\circ$  only. In the case of Galileo 24 satellites constellation this possibility will be the same except for zone  $0-10^\circ$  and latitudes  $40-60^\circ$ .

- The percentage of satellites visible in open area above given angle and the distributions of satellites azimuths for different elevation angle are at different latitudes almost the same.
- The number of BeiDou MEO satellites (27) is greater than the number of Galileo satellites of nominal constellation (24) but orbit altitude of European system (23,222 km) is greater than in China system (21,500 km). That's why the distributions of GDOP coefficient values is for both systems at different latitudes almost the same.
- BeiDou system will meet the demands of China's national security, economic development. One of the most important aims of Galileo is to provide an indigenous alternative high precision positioning system upon which European nations can rely. Galileo is intended to provide horizontal and vertical position measurements within one-metre precision (95%). Both these new SNSs will be committed to providing stable, reliable and quality satellite navigation services for global users.
- Whereas the constellation of 27 BeiDou MEO satellites will provide global coverage at any moment, the constellation of 3 IGSO and 5 GEO satellites will can be used in limited area only. It means that the advantages of D2 NAV message, integrity information in particular will be accessible to users in China and the Asia-Pacific region only.

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