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Supporting Information

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Research paper

Ammonium Dodecahydrododecaborate (NH₄)₂[B₁₂H₁₂]: Hydrogen and Boron Rich Fuel for Jet Propulsion Engines

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Content

Figure S1. (a) DSC and (b) TGA of ADDB

Figure S2. FT-IR spectrum of ADDB

Table S1. Effect of boron and boranes on the flame temperature of JP-10

ISSN 1733-7178; e-ISSN 2353-1843

Supporting Information (SI) for Cent. Eur. J. Energ. Mater. 2022, 19(2): 158-167.

1 Elemental Analysis

Elemental analysis was performed on a VarioMICRO instrument. Elemental analysis showed N 14.74 and H 10.16%, which were close to the theoretical figures, $(N_2H_{20}B_{12}: MW 177.72;$ requires N:15.75% and H: 11.1%). Moreover, no carbonaceous matter was found in the elemental analysis which indicated complete conversion of triethylammonium dodecahydrododecaborate to ADDB.

2 Thermal Analysis

Differential Scanning Calorimetry (DSC), and Thermo-Gravimetric (TGA) analysis of ADDB were done on a PerkinElmer thermal analyzer (STA 6000) under a nitrogen atmosphere at 10°C/min heating rate. Figure S1(a) shows an endothermic peak at 310 °C, of 1497 J/g. The decomposition of ADDB was also reported to be 310 °C by Ivanov *et al.* [15]. This supported the formation of ADDB using the new synthesis route. On the other hand, TGA showed weight gains in two stages (Figure S1(b)). A 15% weight gain was observed in the range 300-600 °C, and a significant weight gain (39%) was seen between 600 and 1000 °C.

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Supporting Information (SI) for Cent. Eur. J. Energ. Mater. 2022, 19(2): 158-167.



(a)

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Supporting Information (SI) for Cent. Eur. J. Energ. Mater. 2022, 19(2): 158-167.



Figure S1. (a) DSC and (b) TGA of ADDB

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Supporting Information (SI) for Cent. Eur. J. Energ. Mater. 2022, 19(2): 158-167.

3 **Spectroscopic Analysis**

Fourier transform infrared spectra (FTIR) were recorded on a Nicolet iS50 FTIR Spectrometer of Thermo Fisher Scientific, which is shown in Figure S2. The FTIR analysis showed strong peaks at:

- $-3231 \text{ cm}^{-1} \text{ (NH4}^{+}\text{)},$
- -2468 cm^{-1} (B–H),
- 1401 cm⁻¹ (N-H), and
- -1070 cm^{-1} (B–B),

which confirmed the functional groups of ADDB.

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Supporting Information (SI) for Cent. Eur. J. Energ. Mater. 2022, 19(2): 158-167.



Figure S2. FT-IR spectrum of ADDB

ISSN 1733-7178; e-ISSN 2353-1843

Supporting Information (SI) for Cent. Eur. J. Energ. Mater. 2022, 19(2): 158-167.

Nuclear magnetic resonance (NMR) spectral analysis of ADDB was carried out in DMSO solution. ¹H and ¹¹B spectra were recorded on a Bruker MSL-300 instrument. ¹H NMR exhibited hydrogen peaks at 7.09 (s, NH₄) and 1.14 (q, B–H) ppm. ¹¹B NMR showed the boron peak at 15 (d, B) ppm.

4 **CEA Details**

Theoretical rocket performance, assuming equilibrium composition during expansion from an infinite area combustor, was computed using the NASA-GLENN Chemical Equilibrium Program (CEA2) [13]. The oxidizer to fuel problem was solved for various ratios and fuels at a fixed equilibrium temperature of 3800 K, an inside pressure of 70 bar and a pressure ratio of 20. The enthalpy from an inbuilt library was used for all ingredients except ADDB and TEDDB. The input file format was as follows.

> Problem: oxidizer (o)/fuel (f) Rocket equilibrium: $t_{cset} = 3800 \text{ K}$ p = 70 bar, $p_{\rm i}/p = 20$ **Ingredient details:** Fuel = JP-10Enthalpy: h = -122.8 kJ/molOxidizer = AirEnthalpy: h = -0.129 kJ/molFuel = ADDBEnthalpy: h = -668.47 kJ/molFuel = TEDDBEnthalpy: h = -3204.73 kJ/molFuel = HTPBEnthalpy: h = -0.26 kJ/molOxidizer = APEnthalpy: h = -295.77 kJ/mol

In case of the JP-10 compositions, the effect of additional fuel i.e. boron and boranes, on the flame temperature was computed and provided as follows.

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Fuel (Composition in %)	Flame temperature [K]
JP-10 (50%)	1103.4
JP-10 (40%) and TEDDB (10%)	1183.2
JP-10 (40%) and ADDB (10%)	1429.9
JP-10 (40%) and boron (10%)	3096.4

Table S1. Effect of boron and boranes on the flame temperature of JP-10

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