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MODELING OF THE STRESS-DEFORMED STATE AND CRACKING INREINFORCED CONCRETE STRUCTURES USING ANSYS MECHANICAL

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A b s t r a c t

The article presents the results of reinforced concrete beams modeling under short-term loading with the use of the ANSYS Mechanical software. The obtained parameters were compared with the experimental ones. The comparison showed a satisfactory concordance of the parameters. In case of the discrepancies, the authors suggest that they are due to the idealization of material properties and structural model of the finite method. The paper is divided into the following chapters: Introduction, Literature Review, Research Significance, Element Types, Real Constants, Material Properties Modeling, Geometric Modeling, Loads and Boundary Conditions, Cracking and Failure and Conclusions. Each of the five conclusions in the last chapter offers a summary of the presented experimental research.

Introduction

Frequently, providing required load bearing capacity of reinforced concrete beams support areas under complex stress-deformed state is a design driver. Nevertheless, behavior and response of these structure elements in case of high level loads (est. $(0.7 - 0.9 V_{ult})$) still remain underexplored. That is why, systematic experimental and theoretical research aimed at improvement of the existing and development of modern computational models of support areas in reinforced concrete elements is of essential importance. Various methods have been used to study the response of structural components. Natural experi-

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ments, while producing real time response, are extremely time consuming and quite costly. As for calculation methods, we must mention, that despite all their efforts (and first of all, price), it is rather unsafe to use them alone for investigation at load levels near 0.9 of ultimate load. Thus, using FEA, such as MacroFe, Ansys, along with natural experiments, is one of the most effective ways to do this task.

In the design of frame concrete structures, it is very important to construct support areas properly. Their combined stress-deformed state makes this task rather difficult. The most common methods used for computational support areas are based on analogies or statistic data and do not describe the real behavior fully.

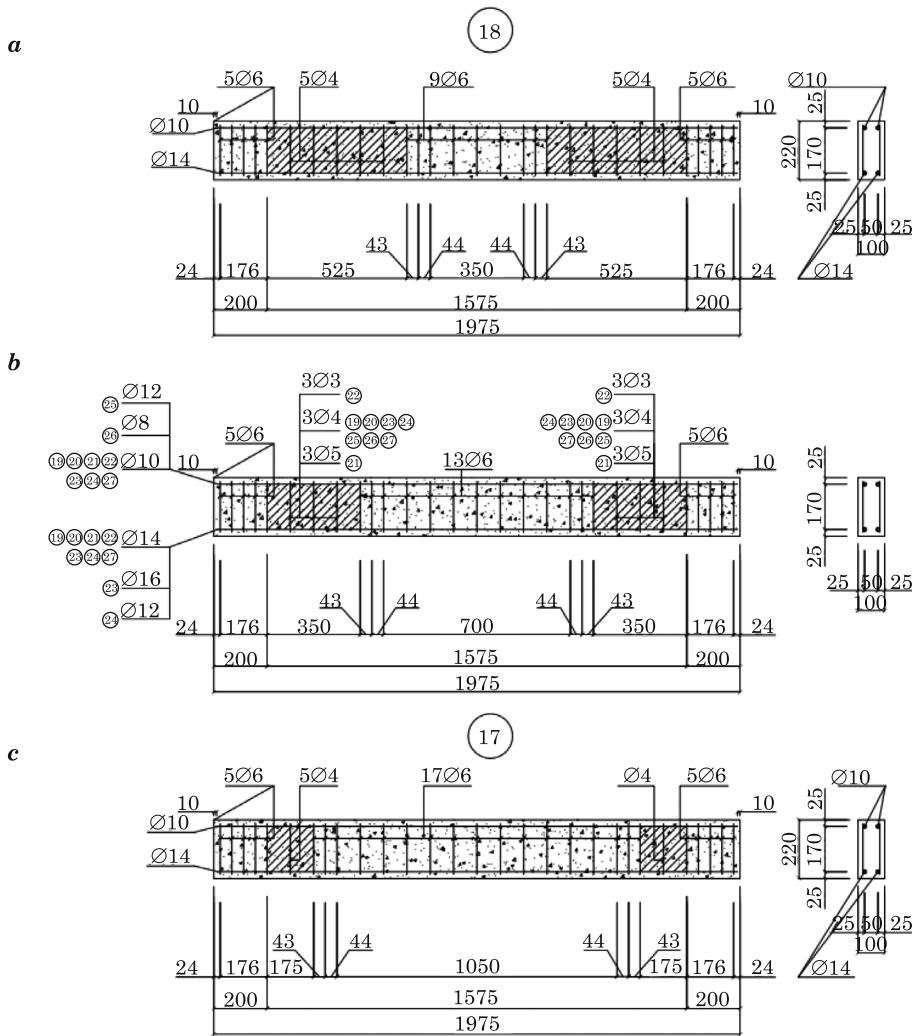
The aim of this experimental research is to study combined stress-deformed state of reinforced concrete beams support areas using finite element modeling with ANSYS Mechanical. Authors intended to investigate possibility of modeling the multifactor experiment, and, at the same time, to focus on the behavior of RC beams support areas, shear cracking, stress distribution in shear and longitude reinforcement under short-term loading.

Material and methods

The data presented herein are basis for the five-factor three-level experimental investigation of reinforced concrete beams support areas under steady loading. Full-sized Hartley-5-type experiment plan includes 27 series. Due to

Table 1
Plan of experiment

Series N	Natural values of the factors				
	a/h_0	C, MPa	Shear reinf. at support sect. [%]	Bottom longt. reinf. [%]	Top longt. reinf. [%]
17	3	C20/25	0.0029 (204 mm)	0.0176 (2Ø14 mm)	0.0090 (2Ø10 mm)
18	1	C20/25	0.0029 (204 mm)	0.0176 (2Ø14 mm)	0.0090 (2Ø10 mm)
19	2	C30/35	0.0029 (204 mm)	0.0176 (2Ø14 mm)	0.0090 (2Ø10 mm)
20	2	C12/15	0.0029 (204 mm)	0.0176 (2Ø14 mm)	0.0090 (2Ø10 mm)
21	2	C20/25	0.0045 (205 mm)	0.0176 (2Ø14 mm)	0.0090 (2Ø10 mm)
22	2	C20/25	0.0016 (203 mm)	0.0176 (2Ø14 mm)	0.0090 (2Ø10 mm)
23	2	C20/25	0.0029 (204 mm)	0.0230 (2Ø16 mm)	0.0090 (2Ø10 mm)
24	2	C20/25	0.0029 (204 mm)	0.0129 (2Ø12 mm)	0.0090 (2Ø10 mm)
25	2	C20/25	0.0029 (204 mm)	0.0176 (2Ø14 mm)	0.0129 (2Ø12 mm)
26	2	C20/25	0.0029 (204 mm)	0.0176 (2Ø14 mm)	0.0058 (2Ø8 mm)
27	2	C20/25	0.0029 (204 mm)	0.0176 (2Ø14 mm)	0.0090 (2Ø10 mm)



Legend:
 – series number

Fig. 1. Beam design: *a* – 18 series, *b* – 19–27 series, *c* – 17 series

considerable complexity of the long-time experiment we decided to use series from 17 to 27 with only one factor variation at one series. Beams of all series are designed to get destroyed along support areas. Experimental plan with a description of experimental factors and variation levels is shown in Table 1.

Construction scheme of authors' experimental RC beam (DOROFEEV et al. 2010, 2012) used as calibration model is shown in Figure 1. Simplified scheme that was used to simulate the experimental beam is shown in Figure 2.

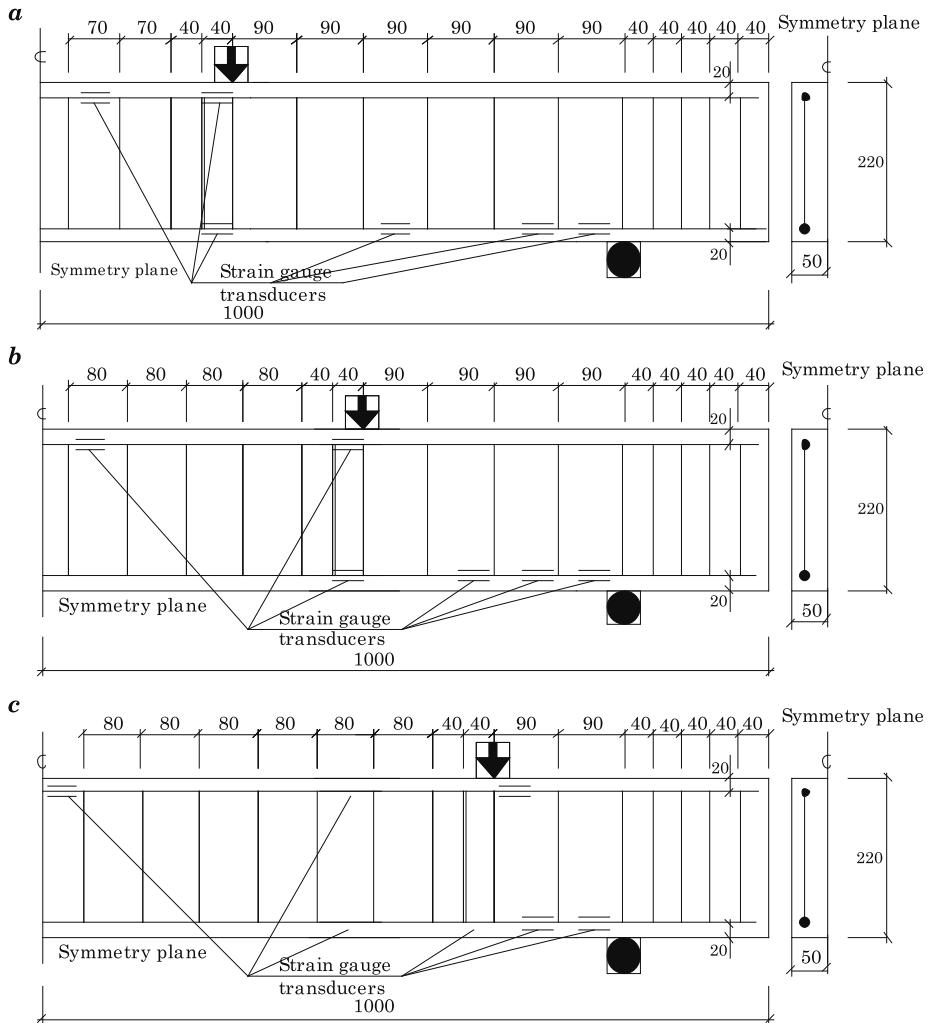


Fig. 2. Model design and strain gauge transducer scheme: *a* – 18 series, *b* – 19–27 series, *c* – 17 series

The element types for this model are shown in Table 2. The Solid 65 element was used to model the concrete. This element is capable of plastic deformation, cracking in three orthogonal directions, crushing, creep and shrinkage. A Solid 45 element was used for steel load plates and supports of the beam. Both these elements have eight nodes with three degrees of freedom at each node. A Link8 element was used to model steel reinforcement. This is 3D spar element and it has two nodes with three degrees of freedom. Also, it is capable of plastic deformation.

Table 2
Elements and real constants used for model

No	Purpose	Element type	Real constants			
			rebar direction	x	y	z
1	concrete	solid 65	reinf. material	0	0	0
			ρ , %	0	0	0
			orient. angle	0	0	0
			orient. angle	0	0	0
			cross-sect. area [m ²]	1.702E-4		
2	bottom longitude reinforcement	link 8	initial strain	0		
3	top longitudereinforcement	link 8	cross-sect. area [m ²]	0.785E-4		
4	shear reinf. in pure flexure zones	link 8	initial strain	0		
5	shear reinf. in support areas	link 8	cross-sect. area [m ²]	0.2827E-4		
			initial strain	0		
			cross-sect. area [m ²]	0.1257E-4		
			initial strain	0		

Real constants for this model are shown in Table 2. Solid 65 elements require constants for rebar assuming a smeared model. In that case, values for Material Number (refers to material model), Volume Ratio and Orientation Angles should be set. Value of zero is entered for all real constants, because discrete reinforcement model is used. For shear, top and bottom reinforcement cross sectional areas and initial strain are defined.

Parameters needed to define material models are shown in Table 3. Complex material model for each element is needed (ANSYS theory reference). The Solid 65 element requires linear isotropic and multilinear isotropic material properties to properly model concrete. The multi-linear isotropic material uses the vonMises failure criterion along with Willam and Warnke model (WILLAM, WARNKE 1974). Modulus of elasticity and Poisson's ratio were assumed according to C20/25 concrete properties. The uniaxial compressive stress-strain relationship for the concrete was obtained using the following equations (DBN – V.2.6-98:2009) with parameters for C20/25 concrete:

$$\sigma_c = f_{(ck)} \sum_{k=1}^5 a_k \left(\frac{\varepsilon}{\varepsilon_{bR}} \right)^k \quad (1)$$

where

- $f_{(ck)}$ – design value of concrete compressive strength;
- a_k – the coefficients of the polynomial depending on the parameters, which are used to describe the stress-strain curve;
- ε – concrete strain deformation;
- ε_{bR} – concrete strain deformation corresponding to the maximum stress value.

Implementation of Willam and Warnke concrete model requires nine constants to be defined:

1. Shear transfer coefficients for an open crack.
2. Shear transfer coefficients for a closed crack.
3. Uniaxial tensile cracking stress.
4. Uniaxial crushing stress (positive).
5. Biaxial crushing stress (positive).
6. Ambient hydrostatic stress state for use with constants 7 and 8.
7. Biaxial crushing stress (positive) under the ambient hydrostatic stress state.
8. Uniaxial crushing stress (positive) under the ambient hydrostatic stress state.
9. Stiffness multiplier for cracked tensile condition.

Concrete failure surface is described with the following equation

$$\frac{1}{z} \frac{\sigma_a}{f_{cu}} + \frac{1}{r(\Theta)} \frac{\tau_a}{f_{cu}} = 1 \quad (2)$$

where:

- f_{cu} – uniaxial compressive strength;
- z – apex of the surface;
- δ_a, τ_a – average stress components;
- Θ – lode angle;
- $r(\Theta)$ – quantity, which indicates the locus of the boundary of the stress surface in the deviatoric stress plane.

The reinforcement elements require linear isotropic and multilinear isotropic material models to be defined. Material of steel load plates and supports of the beam requires only linear isotropic material model.

Concrete, load plates and support element were modeled using volume elements. And the reinforcement was modeled using spar elements. Experimental beam had two symmetry planes and therefore only one quarter of the beam was modeled. Finite element model of concrete and reinforcement with mesh size of 1 cm are shown in Figure 3.

Displacement boundary conditions must be set to constrain the model. To model the symmetry, nodes on the plane must be constrained in the perpendicular direction ($UX = 0$ and $UZ = 0$ constraints). The support was modeled in such a way that a roller was created. A constraint in the UY and UZ (applied as constant values of 0) directions was given to the single line of nodes on the plate. The one tenth of the actual force, F , applied across its centerline at the each node. Main calculation parameters are shown in Table 4.

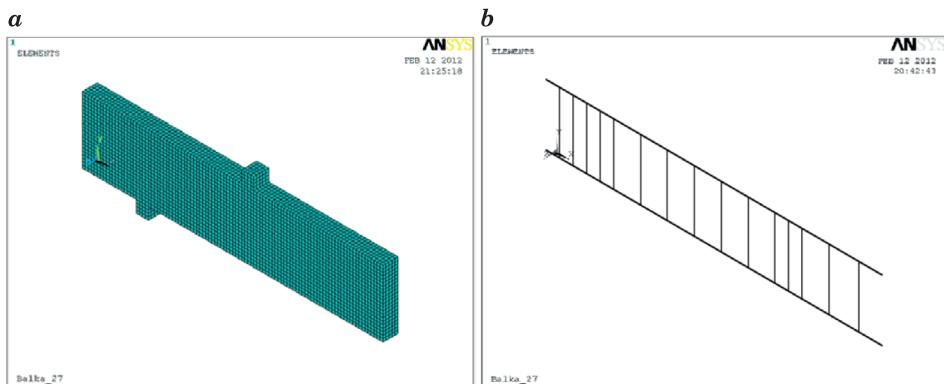


Fig. 3. Mesh of the whole beam (a) mesh of the reinforcement (b)

Table 4
Calculation options

Main parameters	
Analysis type	Static
Displacement control	big displ.
Prestress effects	off
Time at the end of loadstep	P
Number of substeps	10^6
Automatic time stepping	on
Nonlinear algorithm and convergence criteria parameters	
Line search	on
Equiv. Plastic strain	1.0
Full Newton-Rapson option with unsym. matrices	on
Maximum number of iterations	10
Element from control	off
Checking elements	No
Convergence criteria	forces
Tolerance	0.5

Results and Discussion

Crack modeling along with cracking of experimental beam for the most characteristic stages of loading is shown in Figure 4. Ultimate, normal and inclined cracking loads for beam series (17-27) obtained both from natural experiment and modeling using ANSYS are shown in Table 5. We must state, firstly, that cracking load for the model is smaller than for experimental beam;

secondly, that cracks in the model develop more intensively, due to cracking criteria used in Willam-Warnke model. Experiment results indicate that cracking process, both in a simulated beam and corresponding experimental one, begins from the pure flexure zone at the load level close to $0.3 V_{ult}$. Inclined cracks appear in the support areas later, at the loads about. This process is significantly affected by concrete class, percentage of transversal and longitude reinforcement.

Comparing whole cracking pattern in model and experimental beam at the load about $0.8 V_{ult}$ we must state their similarity, as well as equal crack heights. Figures 5 and 6 show the experimental values of midspan deflections and longitude reinforcement deformations of the series 22 and the results of their simulations, where 22-1(1) means "22 series, 1st beam, left side of experimen-

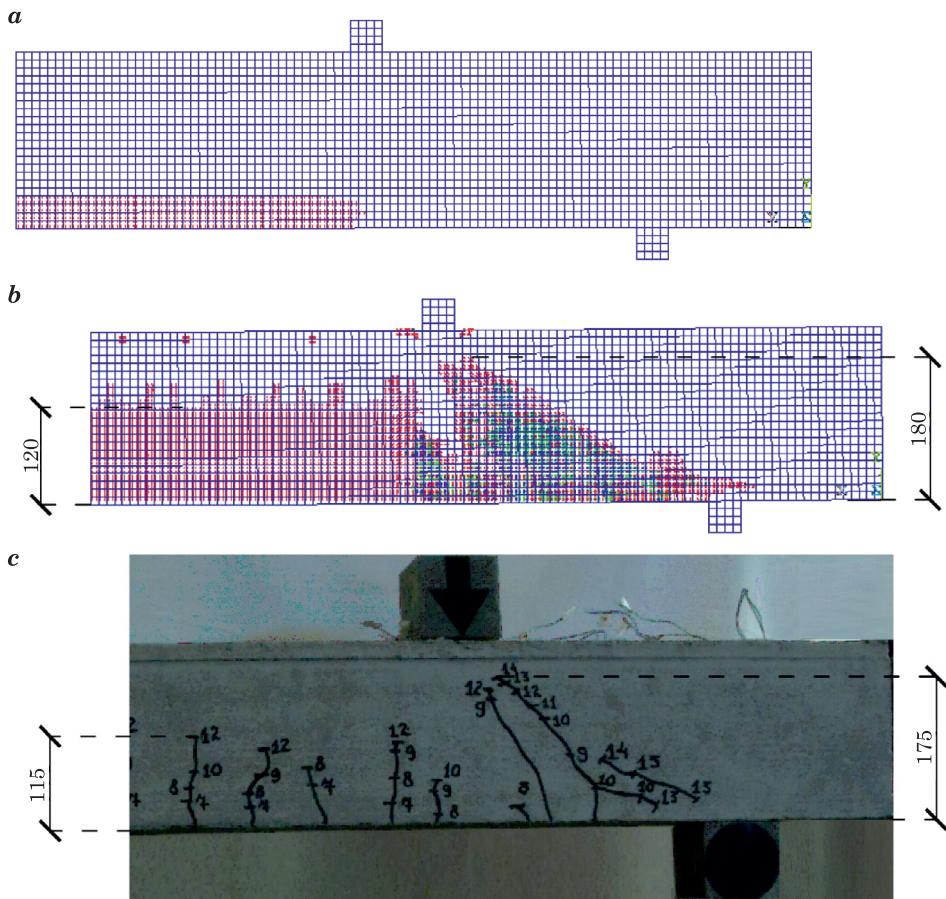


Fig. 4. Modeling of crack at load levels: $0.3 V_{ult}$ (a), $0.8 V_{ult}$ (b); appearance and crack opening in the experimental sample (c)

tal beam" and 22-1(2) means "22 series, 1st beam, right side of experimental beam" respectively. It is obvious, that up to, congruence is more than satisfactory. Further increasing discrepancy is due to method limitations.

Table 5
Failure and cracking loads

Series	Model			Experiment		
	V_{ult} [kN]	$M_{(cr)}$ (normal cracks at pureflexure zone) [kNm]	$F_{(cr)}$ (inclined cracks at the support area) [kN]	V_{ult} [kN]	$M_{(cr)}$ (normal cracks at pureflexure zone) [kNm]	$F_{(cr)}$ (inclined cracks at the support area) [kN]
0-17	147	18	20	140	20	30
0-18	60	46	50	64	50	70
0-19	120	26	27	116	27	65
0-20	75	15	28	72	28	45
0-21	86	25	33	96	33	48
0-22	81	22	32	82	32	40
0-23	96	25	33	100	33	48
0-24	65	21	26	72	26	48
0-25	86	25	30	89	30	48
0-26	86	23	30	84	30	40
0-27	86	23	30	86	30	48

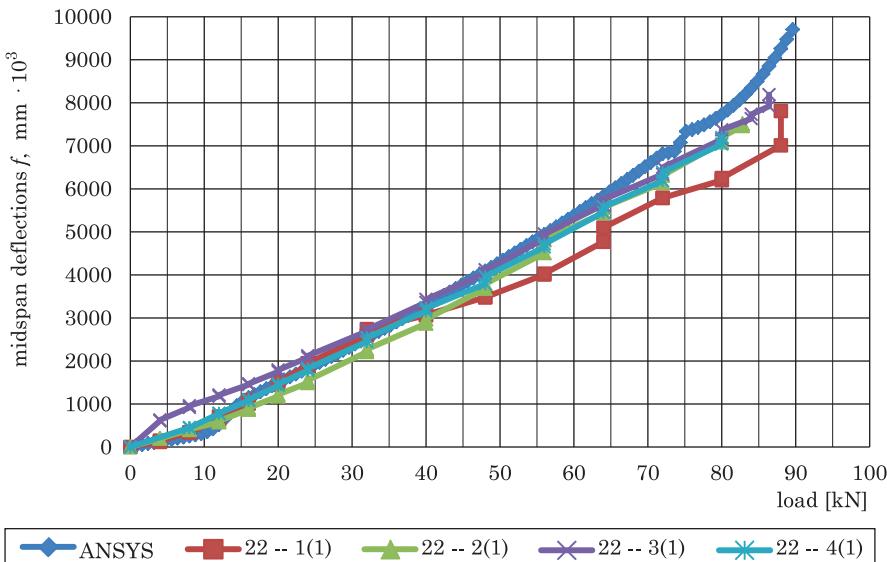


Fig. 5. Calculated and experimental values of beam midspan deflections

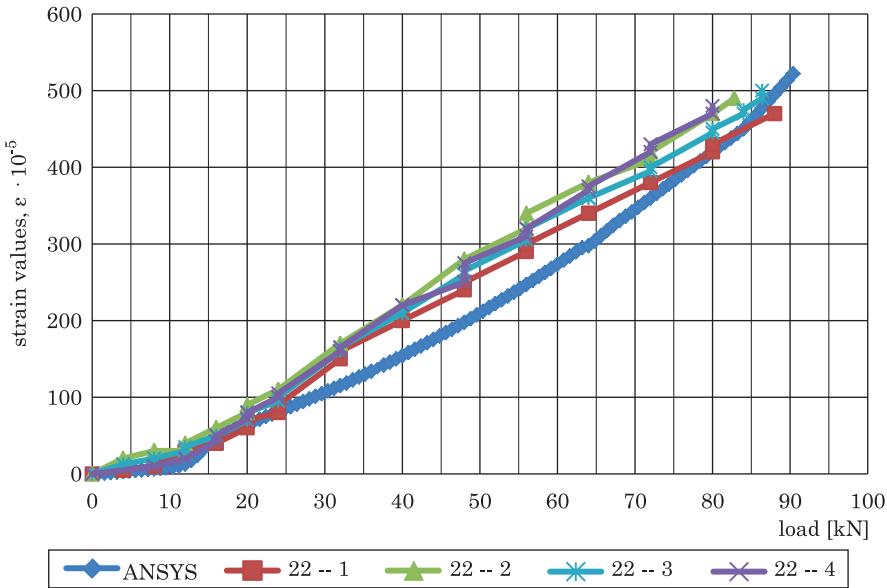


Fig. 6. Calculated and experimental strain values in bottom rebar sat mid span

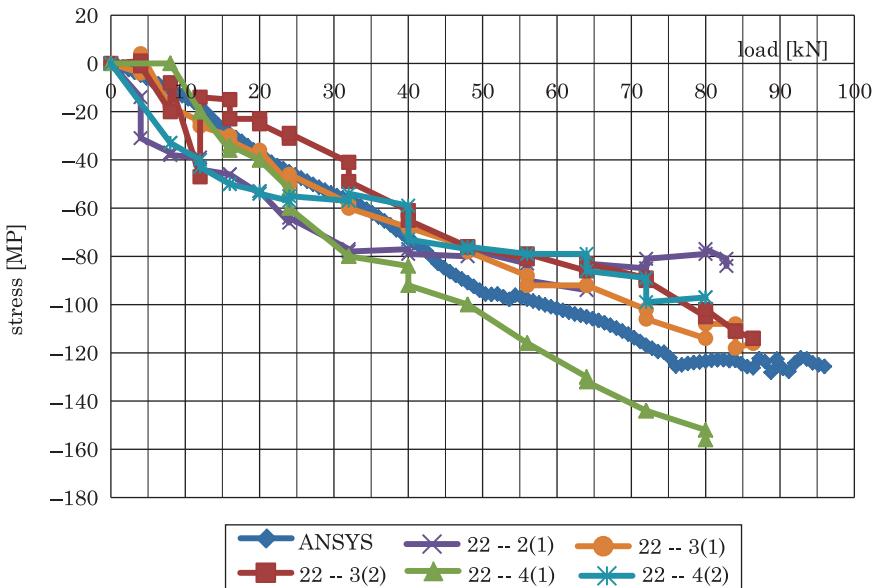


Fig. 7. Calculated and experimental stress values in top reinforcement near the load plate (at the end of support area)

Calculated values of longitude reinforcement stresses along with modeled values are shown in Figures 7–10. Experimental values have been obtained by processing the strain gauge transducer chain data, located on bottom and top longitude reinforcement both in pure flexure zone and support areas. It is fair to say that, in total, congruence of experimental and modeled stress-deformed state of top reinforcement is stable (Fig. 6 and 7).

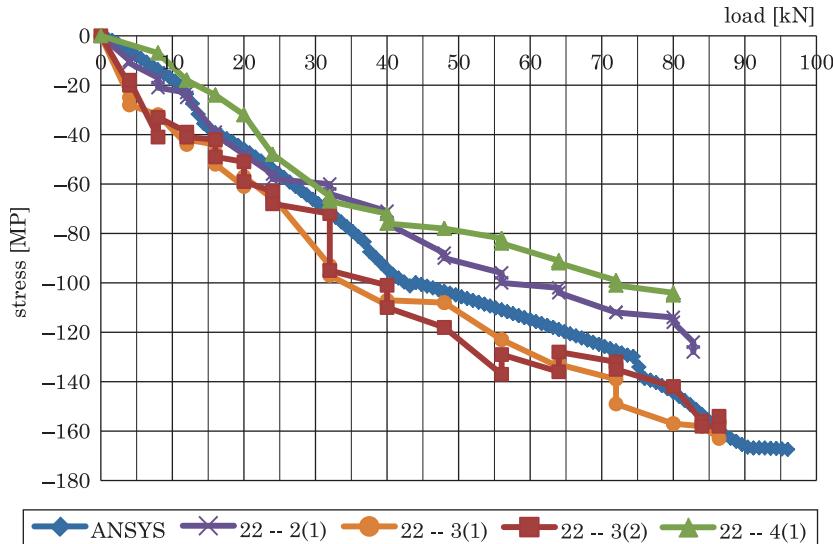


Fig. 8. Calculated and experimental stress values in top reinforcement in pure flexure zone

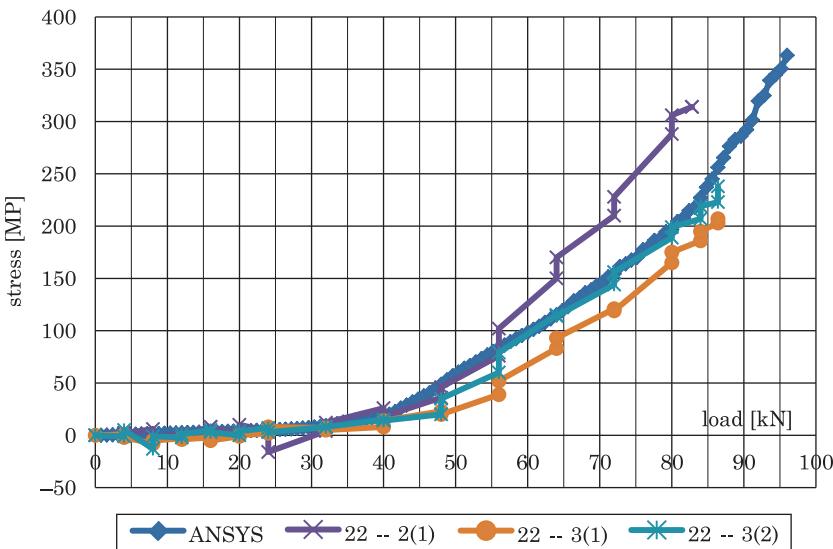


Fig. 9. Calculated and experimental stress values in bottom reinforcement near the support hinge

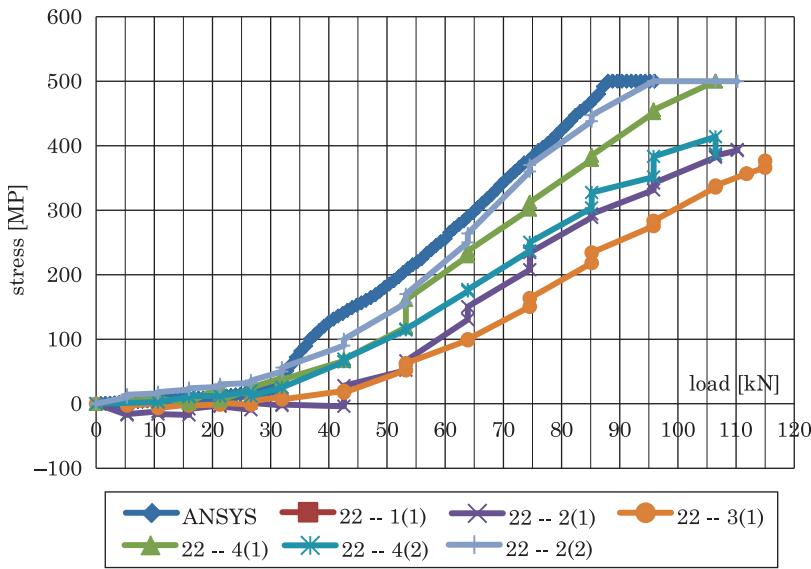


Fig. 10. Calculated and experimental stress values in bottom reinforcement at a point located at a distance of 9 cm from the support (at the mouth of the critical inclined crack)

As for bottom reinforcement, they differ more significantly. In a greater degree this concerns pure flexure zone on loading, which is close to failure. Under these load levels cracking process in model is like an avalanche, leading to substantial strain growth. To a smaller extent it concerns bottom reinforcement in support areas, although model strain rate is bigger than in natural beams. And, vice versa, modeled top reinforcement strain is less than experimental ones.

Stressed state of shear reinforcement was studied according to scheme, shown in Figure 11. Relations between load and modelled stresses in shear reinforcement bars in the support areas are shown in Figures 12, 13, 14.

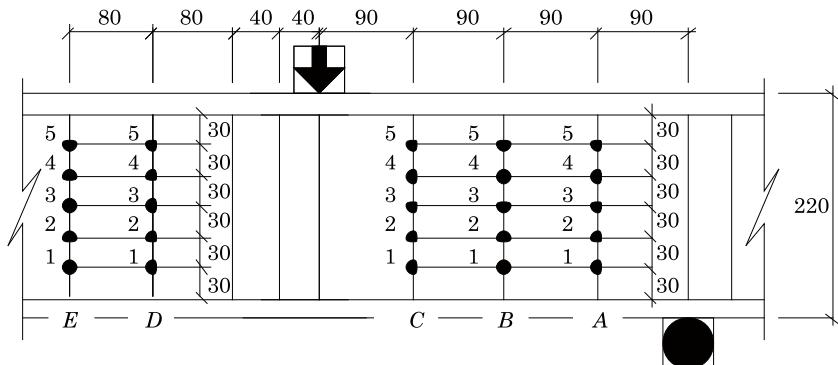


Fig. 11. Characteristic points of the shear reinforcement in the bars A, B, C in the support areas, D, E – in pure flexure zone

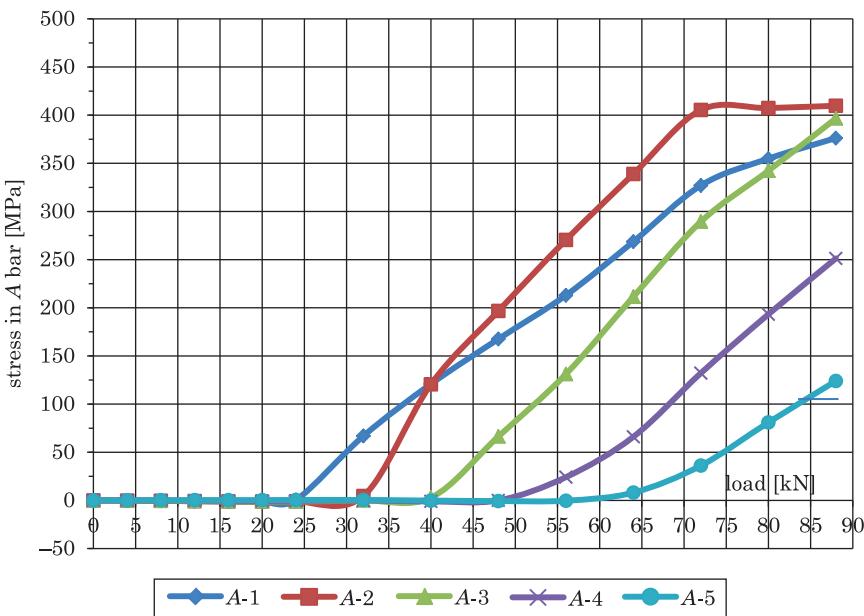


Fig. 12. Force-stress graph for the charecteristic points of the rebar A

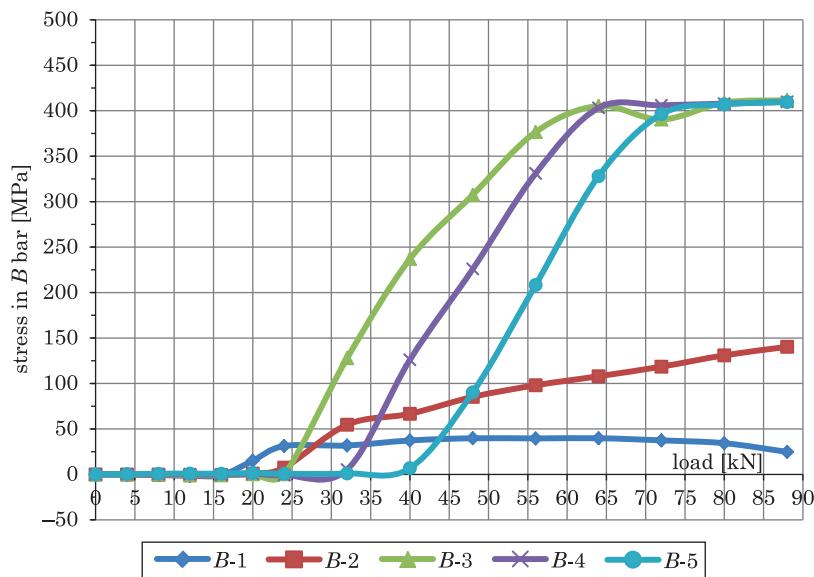


Fig. 13. Force-stress graph for the charecteristic points of the rebar B

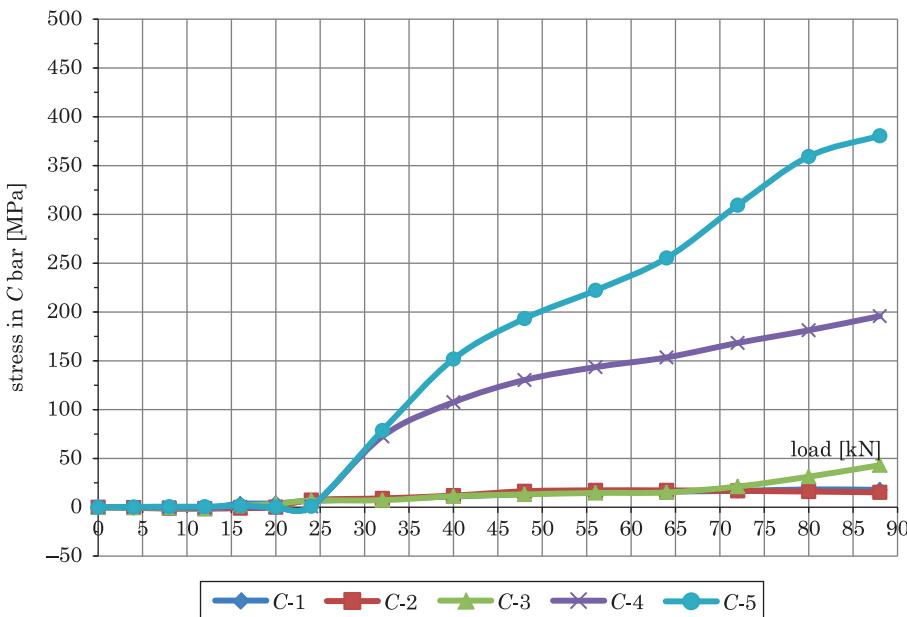


Fig. 14. Force-stress graph for the charecteristic points of the rebar *C*

As we can see, results shown in Figure 12, indicate that stresses in shear rebar *A* are unevenly distributed. Initially, the lowest point 1 is “turned on”, then point 2, etc. Stresses in 5th point begin to grow suginificantly only with load level about $0.75 V_{ult}$. The most stressed point in *A* shear rebar is point 2. At load level about $0.80 V_{ult}$, stresses in that point reach the yield strength. Point 1 is the second most stressed point, point 3 – the third one – its stresses reach yield strength at the time of failure.

Shear rebar *B*, which is located near midspan (Figure 13), is the most heavily loaded, as stresses reach yield strength in 3 of 5 of his characteristic points. It should be noted, that in these points stresses reach yield strength at load level about. Ultimate stresses in points 1, 2 reaches only 30 and 140 MPa respectively.

Shear reinforcement bar *C*, adjacent to the pure flexure zone, is the least loaded. Almost the entire height (points 1, 2, 3) stresses in their points are less than 50 MPa. Point 4 stress reaches 200 MPa. The most heavily loaded point in this rebar is point 5 (380 MPa), which is very close to the yield point. Modeling results which are shown in Figures 12–14, allow to indicate the direction and area of probable cracks occurrence (Figure 15). In order to do that we can draw a line across points in shear bars where stresses reach ultimate values. Cracking in the support area of experimental beam along with the first principal strain color map is shown in Figure 16. The results presented here indicate that modeled crack patterns practically coincide with the experimental ones.

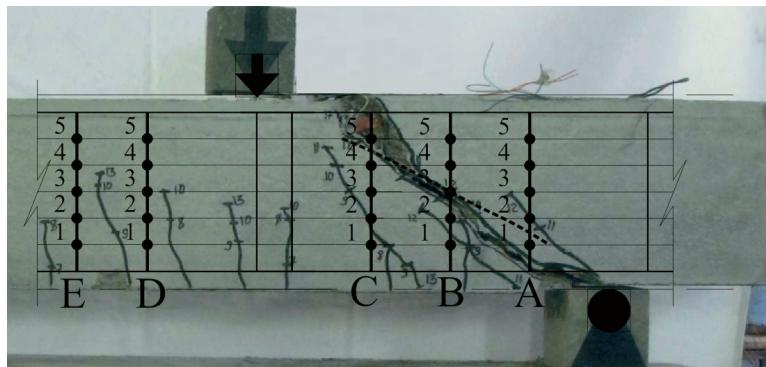


Fig. 15. Cracking and failure of experimental beam support area

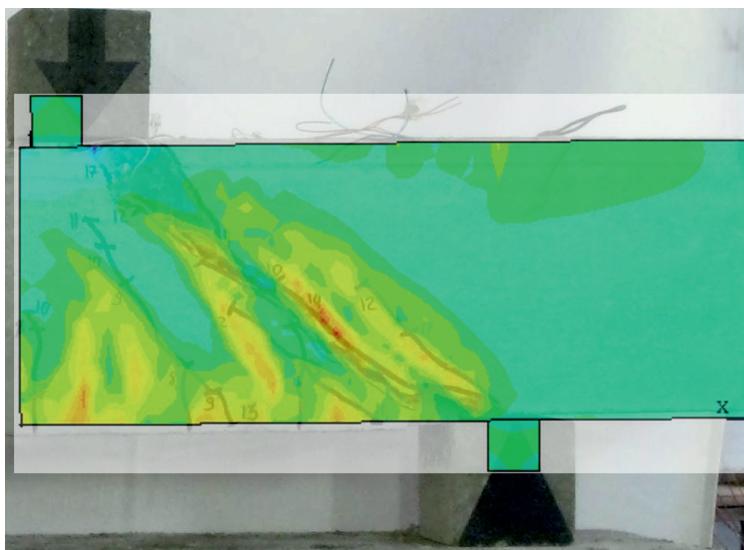


Fig. 16. Cracking in the support area along with the first principal strain color map

Conclusions

1. Failure mechanism modeling using FEA Ansys shows reliable results while received failure load is very close to experimental.
2. Modeled cracking load is also close to experimental. At the same time, it should be noted that cracking pattern in model is rather simplified compared to natural beams and crack growth rate in model is greater than in experimental beam due to restrictions of Willam-Warnke concrete model used in Ansys.
3. Modeled deflections and strains are almost the same as the experimental ones up to.

4. The stresses in top and bottom reinforcement bars received through modeling showed very good agreement and confirmed by direct measurements, actual cracking process in the test beams, as well as other experimental parameters. Discrepancies are due to the idealization of material properties and structural model of the finite element method.

5. Modeling of shear reinforcement provided that good agreement according to other parameters of stress-deformed state can provide valuable data on the growth and change in the pattern of stresses in the shear reinforcement bars, which are difficult to get with a direct experiment. These data are indirectly confirmed as the development of the main crack in the experimental beam.

6. Modeling of the stress-deformed state of researched elements using FEA ANSYS unlike other approved local software systems (in particular, "Lyra", "SCAD", "Monomakh") gives an opportunity to predict the formation and development of both normal and inclined cracks on support areas of bending elements, and the possible mechanism of their failure.

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