On fault tolerant control structures incorporating fault estimation

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The paper provides the minimal necessary modifications of linear matrix inequality conditions for the mixed H_2/H_{∞} control design as well as for the augmented observer-based fault estimation to be mutually compatible in joint design of integrated fault estimation and fault tolerant control. To be possible, within this integration, to design the controller which guarantees a pre-specified H_{∞} norm disturbance attenuation level, the design conditions has to be regularized using the H_2 performance index and, moreover, augmented fault observer must be of enforced dynamics. Analyzing the ambit of performances given on the mixed H_2/H_{∞} design, the joint design conditions are formulated as a minimization problem subject to convex constraints expressed by a system of LMIs. The feasibility of the conditions is demonstrated by a numerical example.

Key words: linear systems, fault tolerant control, fault estimation, linear matrix inequalities, H_{∞} norm, H_2/H_{∞} control strategy.

1. Introduction

A model-based fault tolerant control (FTC) can be realized as control-laws set dependent, exploiting fault detection and isolation decision to reconfigure the control structure or as fault estimation dependent, preferring fault compensation within robust control framework. Whilst integration of FTC with the fault localization decision technique requires a selection of optimal residual thresholds as well as a robust and stable reconfiguration mechanism [5], the fault estimation dependent FTC structures eliminate a threshold subjectivism and integrate FTC and estimation problems into one robust optimization task [17], [18].

The H₂-norm is one of the most important characteristics for linear time-invariant control systems, and is often used as performance index of the system in analysis and

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design problems. Thus, problems concerning H_2 , as well as H_{∞} control have been studied by many authors [9], [16], [20], [21]. Adding H_2 objective to H_{∞} control design, a mixed H_2/H_{∞} control problem was formulated in [13], with the goal to design the gain matrix of the state-feedback control law such that minimizes H_2 norm subject the constraint on H_{∞} norm of the disturbance transfer function. Such integrated design strategy corresponds to optimization of the design parameters to satisfy desired specifications and to optimize the performance of the closed-loop system. Because of the importance of control systems with these properties, considerable attention was dedicated to mixed H_2/H_{∞} closed-loop performance criterion in design [1], [7], [10], [23], as well as to formulate the LMI-based computational technique [19], [24] to solve them or to exploit non-linear multi-objective algorithms for non-smooth optimization in this design task [11].

The approach, in which faults estimates are used in the control structure to compensate the effects of acting faults, is adopted in modern FTC techniques [3], [8], [28]. Integrated single-step methods of fault estimation and FTC, for linear systems subject to bounded actuator or sensor faults, are proposed in [17], [27]. Specified via LMI formulations and solved using H_{∞} or mixed H_2/H_{∞} optimization, the observer structures are augmented fault state observers in the standard Luenberger form [22] or unknown input augmented fault observers [25]. To guarantee desired time response, an LMI regional pole placement design strategy is proposed in [26], [27]. However, such formulation introduces minimally two additive LMIs, which increase conservatism of the design conditions. Moreover, due to the extended structure of the parameter matrices of the fault tolerant controller [17], the LMI-based H_{∞} control design solutions are, in general, marginal feasible.

Re-modifying the results given in [4] by an updated \mathcal{D} -stability circle criterion [14], the LMI conditions for fault observer design, proposed in this paper, inherently includes within minimal set of LMIs the regional pole placement condition to guarantee a suitable fault observer dynamics and, in consequence, a satisfactory time response of the FTC structure. Moreover, formulating as a problem subject to convex constraints, a mixed H₂/H_{∞} standard design method presented in [20] is modified relative the dual property of Schur complements. Accordingly, since an extended Lyapunov function is exploited, the obtained H₂/H_{∞} design conditions are regularized under acting of H₂ constraint.

The content and scope of the paper are as follows. Placed after the introduction, presented in Sec. 1, the problem formulation and the basic preliminaries are given in Sec. 2 Next, Sec. 3 recalls the formulation of the fault-augmented observer for continuous-time linear systems and the controller design conditions in the framework of LMIs are recast in Sec. 4 Then, in Sec. 5, in response to fault compensation principle for such type of fault observers, the design conditions for the fault tolerant tracking control structures are derived, reflecting the mixed H_2/H_{∞} control idea. The relevance of the proposed approach is illustrated by a numerical example in Sec. 6 and Sec. 7 draws some concluding remarks.

Throughout the paper, the notations is narrowly standard in such a way that \mathbf{x}^T , \mathbf{X}^T denotes the transpose of the vector \mathbf{x} and matrix \mathbf{X} , respectively, $diag(\cdot)$ denotes a block diagonal matrix, $rank(\cdot)$ remits the rank of a matrix, for a square matrix $\mathbf{X} < 0$ means

that **X** is a symmetric negative definite matrix, the symbol I_n indicates the *n*-th order unit matrix, \Re denotes the set of real numbers and $\Re^{n \times r}$ refers to the set of all $n \times r$ real matrices.

2. Basic preliminaries

The considered systems are described in the state-space form by the set of equations

$$\dot{\boldsymbol{q}}(t) = \boldsymbol{A}\boldsymbol{q}(t) + \boldsymbol{B}\boldsymbol{u}(t) + \boldsymbol{F}\boldsymbol{f}(t), \qquad (1)$$

$$\mathbf{y}(t) = \boldsymbol{C}\boldsymbol{q}(t), \qquad (2)$$

where $\boldsymbol{q}(t) \in \mathfrak{R}^n$, $\boldsymbol{u}(t) \in \mathfrak{R}^r$, $\boldsymbol{y}(t) \in \mathfrak{R}^m$ are the vectors of the state, input and output variables, $\boldsymbol{f}(t) \in \mathfrak{R}^p$ is the fault vector, $\boldsymbol{A} \in \mathfrak{R}^{n \times n}$, $\boldsymbol{B} \in \mathfrak{R}^{n \times r}$, $\boldsymbol{C} \in \mathfrak{R}^{m \times n}$, $\boldsymbol{F} \in \mathfrak{R}^{n \times p}$ are real finite values matrices, m, r, p < n and

$$\operatorname{rank}\left[\begin{array}{cc} \mathbf{A} & \mathbf{F} \\ \mathbf{C} & \mathbf{0} \end{array}\right] = n + p. \tag{3}$$

The transfer function matrix with respect to (1), (2) is

$$\boldsymbol{G}(s) = \boldsymbol{C}(s\boldsymbol{I}_n - \boldsymbol{A})^{-1}\boldsymbol{B}, \qquad (4)$$

which gives the relationship to the state representation.

It is assumed that the fault f(t) may occur at an uncertain time, is slowly-varying and bounded and, to estimate such faults, it is supposed that the pair (A, C) is observable.

In order to analyze whether a system is stable under defined quadratic constraints, the concept can be summarized by the following LMI forms.

Lemma 1 [6] The matrix **A** is Hurwitz and $\|\mathbf{G}(s)\|_2 < \gamma_2$ if there exists a symmetric positive definite matrix $\mathbf{V} \in \Re^{n \times n}$ and a positive scalar $\gamma_2 \in \Re$, such that

$$\boldsymbol{V} = \boldsymbol{V}^T > 0, \tag{5}$$

$$\boldsymbol{A}\boldsymbol{V} + \boldsymbol{V}\boldsymbol{A}^T + \boldsymbol{B}\boldsymbol{B}^T < 0, \qquad (6)$$

$$tr(\boldsymbol{C}\boldsymbol{V}\boldsymbol{C}^{T}) < \gamma_{2}^{2}, \tag{7}$$

where $\gamma_2 > 0$, $\gamma_2 \in \Re$ is H_2 norm of the transfer function matrix (4) of the system.

Lemma 2 [2] The matrix \mathbf{A} is Hurwitz and $\|\mathbf{G}(s)\|_{\infty} < \gamma_{\infty}$ if there exists a symmetric positive definite matrix $\mathbf{R} \in \Re^{n \times n}$ and a positive scalar $\gamma_{\infty} \in \Re$ such that

$$\boldsymbol{R} = \boldsymbol{R}^T > 0, \tag{8}$$

$$\begin{bmatrix} \mathbf{A}^{T}\mathbf{R} + \mathbf{R}\mathbf{A} & * & * \\ \mathbf{B}^{T}\mathbf{R} & -\gamma_{\infty}\mathbf{I}_{r} & * \\ \mathbf{C} & \mathbf{0} & -\gamma_{\infty}\mathbf{I}_{m} \end{bmatrix} < 0, \qquad (9)$$

where $I_r \in \Re^{r \times r}$, $I_m \in \Re^{m \times m}$ are identity matrices and $\gamma_{\infty} > 0$, $\gamma_{\infty} \in \Re$ is H_{∞} norm of the transfer function matrix (4) of the system.

Hereafter, * denotes the symmetric item in a symmetric matrix.

Lemma 3 [14] *The matrix* **A** *is D*-*stable Hurwitz if for given positive scalars* $a, \rho \in \Re$, $a > \rho > 0$, there exists a symmetric positive definite matrix $P \in \Re^{n \times n}$ such that

$$\begin{bmatrix} -\rho \boldsymbol{P} & \ast \\ \boldsymbol{P}\boldsymbol{A} + a\boldsymbol{P} & -\rho \boldsymbol{P} \end{bmatrix} < 0,$$
(10)

while the eigenvalues of **A** are clustered in the circle with the origin $s_o = (-a+0i)$ and radius ρ within the complex plane S.

3. Observer-based fault estimation

Limiting to time invariant systems and focusing on fault estimation for slowlyvarying faults, the fault observer is considered in the following form

$$\dot{\boldsymbol{q}}_{e}(t) = \boldsymbol{A}\boldsymbol{q}_{e}(t) + \boldsymbol{B}\boldsymbol{u}(t) + \boldsymbol{F}\boldsymbol{f}_{e}(t) + \boldsymbol{J}(\boldsymbol{y}(t) - \boldsymbol{y}_{e}(t)), \qquad (11)$$

$$\mathbf{y}_e(t) = \mathbf{C} \mathbf{q}_e(t), \qquad (12)$$

where $\boldsymbol{q}_e(t) \in \Re^n$, $\boldsymbol{y}_e(t) \in \Re^m$, $\boldsymbol{f}_e(t) \in \Re^p$ are estimates of the system states vector, the output variables vector and the fault vector, respectively, and $\boldsymbol{J} \in \Re^{n \times m}$ is the observer gain matrix.

If the observer errors between the system state vector and the observer state vector, as well as between the fault vector and the vector of its estimate, are defined as follows

$$\boldsymbol{e}_q(t) = \boldsymbol{q}(t) - \boldsymbol{q}_e(t), \qquad \boldsymbol{e}_f(t) = \boldsymbol{f}(t) - \boldsymbol{f}_e(t), \tag{13}$$

it is reasonable to consider for slowly-varying faults [27]

$$\dot{\boldsymbol{f}}(t) \approx \boldsymbol{0}, \qquad \dot{\boldsymbol{f}}_e(t) = \boldsymbol{L}\boldsymbol{C}\boldsymbol{e}_q(t),$$
(14)

where $L \in \Re^{p \times m}$ is the design parameter. The goal is to synthesize the couple (J, L) in such a way that the fault observer (11)–(14) is stable.

In order to be able to formulate the fault observer design conditions, (1), (2) and (14) are rewritten compositely as

$$\begin{bmatrix} \dot{\boldsymbol{q}}(t) \\ \dot{\boldsymbol{f}}(t) \end{bmatrix} = \begin{bmatrix} \boldsymbol{A} & \boldsymbol{F} \\ \boldsymbol{0} & \boldsymbol{0} \end{bmatrix} \begin{bmatrix} \boldsymbol{q}(t) \\ \boldsymbol{f}(t) \end{bmatrix} + \begin{bmatrix} \boldsymbol{B} \\ \boldsymbol{0} \end{bmatrix} \boldsymbol{u}(t), \quad (15)$$

$$\mathbf{y}(t) = \begin{bmatrix} \mathbf{C} & \mathbf{0} \end{bmatrix} \begin{bmatrix} \mathbf{q}(t) \\ \mathbf{f}(t) \end{bmatrix}$$
(16)

and, analogously, (11), (12), (14) as

$$\begin{bmatrix} \dot{\boldsymbol{q}}_{e}(t) \\ \dot{\boldsymbol{f}}_{e}(t) \end{bmatrix} = \begin{bmatrix} \boldsymbol{A} & \boldsymbol{F} \\ \boldsymbol{0} & \boldsymbol{0} \end{bmatrix} \begin{bmatrix} \boldsymbol{q}_{e}(t) \\ \boldsymbol{f}_{e}(t) \end{bmatrix} + \begin{bmatrix} \boldsymbol{B} \\ \boldsymbol{0} \end{bmatrix} \boldsymbol{u}(t) + \begin{bmatrix} \boldsymbol{J} \\ \boldsymbol{L} \end{bmatrix} \boldsymbol{C}\boldsymbol{e}_{q}(t), \quad (17)$$

$$\mathbf{y}_{e}(t) = \begin{bmatrix} \mathbf{C} & \mathbf{0} \end{bmatrix} \begin{bmatrix} \mathbf{q}_{e}(t) \\ \mathbf{f}_{e}(t) \end{bmatrix}.$$
 (18)

Thus, introducing the notations

$$\boldsymbol{q}^{\circ T}(t) = \begin{bmatrix} \boldsymbol{q}^{T}(t) & \boldsymbol{f}^{T}(t) \end{bmatrix}, \qquad \boldsymbol{q}_{e}^{\circ T}(t) = \begin{bmatrix} \boldsymbol{q}_{e}^{T}(t) & \boldsymbol{f}_{e}^{T}(t) \end{bmatrix},$$
(19)

$$\boldsymbol{A}^{\circ} = \begin{bmatrix} \boldsymbol{A} & \boldsymbol{F} \\ \boldsymbol{0} & \boldsymbol{0} \end{bmatrix}, \ \boldsymbol{B}^{\circ} = \begin{bmatrix} \boldsymbol{B} \\ \boldsymbol{0} \end{bmatrix}, \ \boldsymbol{J}^{\circ} = \begin{bmatrix} \boldsymbol{J} \\ \boldsymbol{L} \end{bmatrix}, \ \boldsymbol{C}^{\circ} = \begin{bmatrix} \boldsymbol{C} & \boldsymbol{0} \end{bmatrix},$$
(20)

where $\mathbf{A}^{\circ} \in \Re^{(n+p) \times (n+p)}$, $\mathbf{J}^{\circ} \in \Re^{(n+p) \times m}$, $\mathbf{C}^{\circ} \in \Re^{m \times (n+p)}$, $\mathbf{B}^{\circ} \in \Re^{(n+p) \times r}$, $\mathbf{q}^{\circ}(t)$, $\mathbf{q}_{e}^{\circ}(t) \in \Re^{n+p}$, then from the above follows

$$\dot{\boldsymbol{q}}^{\circ}(t) = \boldsymbol{A}^{\circ} \boldsymbol{q}^{\circ}(t) + \boldsymbol{B}^{\circ} \boldsymbol{u}(t), \qquad (21)$$

$$\dot{\boldsymbol{q}}_{e}^{\circ}(t) = \boldsymbol{A}^{\circ}\boldsymbol{q}_{e}^{\circ}(t) + \boldsymbol{B}^{\circ}\boldsymbol{u}(t) + \boldsymbol{J}^{\circ}(\boldsymbol{y}(t) - \boldsymbol{y}_{e}(t)), \qquad (22)$$

$$\mathbf{y}(t) = \mathbf{C}^{\circ} \mathbf{q}^{\circ}(t), \qquad \mathbf{y}_{e}(t) = \mathbf{C}^{\circ} \mathbf{q}_{e}^{\circ}(t).$$
(23)

This leads to the following equation

$$\dot{\boldsymbol{e}}^{\circ}(t) = (\boldsymbol{A}^{\circ} - \boldsymbol{J}^{\circ}\boldsymbol{C}^{\circ})\boldsymbol{e}^{\circ}(t) = \boldsymbol{A}_{e}^{\circ}\boldsymbol{e}^{\circ}(t), \qquad (24)$$

where

$$\boldsymbol{A}_{e}^{\circ} = \boldsymbol{A}^{\circ} - \boldsymbol{J}^{\circ} \boldsymbol{C}^{\circ}, \qquad (25)$$

$$\boldsymbol{e}^{\circ}(t) = \boldsymbol{q}^{\circ}(t) - \boldsymbol{q}^{\circ}_{\boldsymbol{e}}(t), \qquad (26)$$

 $q_e^{\circ}(t)$ and $y_e(t)$ represent the estimates of $q^{\circ}(t)$ and y(t), respectively, and $e^{\circ}(t)$ is the fault observer error.

4. Control design strategies

It is assumed that the system (22), (23) is controllable to be controlled by the state feedback control law

$$\boldsymbol{u}(t) = -\boldsymbol{K}^{\circ}\boldsymbol{q}^{\circ}(t), \qquad (27)$$

$$\boldsymbol{K}^{\circ} = \left[\begin{array}{cc} \boldsymbol{K}_{q} & \boldsymbol{K}_{f} \end{array} \right], \tag{28}$$

where $\mathbf{K}^{\circ} \in \Re^{r \times (n+p)}$ is the control law gain matrix.

If the system set-point may vary under normal circumstances, it is desired to adjust the plant working point indirectly by the output following. That this be done, the difference between the output $\mathbf{y}(t)$ and its desired value $\mathbf{w}(t)$ has to be exploited so that the outputs could try to follow their desired values. It is possible to ensure this idea by the integral tracking, where the controller integral term minimized this difference. In order to eliminate tracking error, the control law (27) is extended by an integral component of the form [29]

$$\boldsymbol{e}_{w}(t) = \int_{0}^{t} (\boldsymbol{w}(\tau) - \boldsymbol{y}(\tau)) \mathrm{d}\tau, \qquad (29)$$

which is joined to (27) in such a way that

$$\boldsymbol{u}(t) = -\boldsymbol{K}^{\bullet}\boldsymbol{q}^{\bullet}(t), \qquad (30)$$

where

$$\boldsymbol{q}^{\bullet T}(t) = \begin{bmatrix} \boldsymbol{q}^{\circ T}(t) & \boldsymbol{e}_{w}^{T}(t) \end{bmatrix} = \begin{bmatrix} \boldsymbol{q}^{T}(t) & \boldsymbol{f}^{T}(t) & \boldsymbol{e}_{w}^{T}(t) \end{bmatrix}, \quad (31)$$

$$\boldsymbol{K}^{\bullet} = \begin{bmatrix} \boldsymbol{K}^{\circ} & \boldsymbol{K}_{w} \end{bmatrix} = \begin{bmatrix} \boldsymbol{K}_{q} & \boldsymbol{K}_{f} & \boldsymbol{K}_{w} \end{bmatrix}, \qquad (32)$$

while $\boldsymbol{q}^{\bullet}(t) \in \Re^{n+p+m}, \boldsymbol{K}^{\bullet} \in \Re^{r \times (n+p+m)}$.

From (29) follows directly

$$\dot{\boldsymbol{e}}_{w}(t) = \boldsymbol{w}(t) - \boldsymbol{y}(t) \tag{33}$$

and the system model (15), (16) has to be expanded as

$$\begin{bmatrix} \dot{\boldsymbol{q}}(t) \\ \dot{\boldsymbol{f}}(t) \\ \dot{\boldsymbol{e}}_{w}(t) \end{bmatrix} = \begin{bmatrix} \boldsymbol{A} & \boldsymbol{F} & \boldsymbol{0} \\ \boldsymbol{0} & \boldsymbol{0} & \boldsymbol{0} \\ -\boldsymbol{C} & \boldsymbol{0} & \boldsymbol{0} \end{bmatrix} \begin{bmatrix} \boldsymbol{q}(t) \\ \boldsymbol{f}(t) \\ \boldsymbol{e}_{w}(t) \end{bmatrix} + \begin{bmatrix} \boldsymbol{B} \\ \boldsymbol{0} \\ \boldsymbol{0} \end{bmatrix} \boldsymbol{u}(t) + \begin{bmatrix} \boldsymbol{0} \\ \boldsymbol{0} \\ \boldsymbol{I}_{m} \end{bmatrix} \boldsymbol{w}(t), \quad (34)$$
$$\boldsymbol{y}(t) = \begin{bmatrix} \boldsymbol{C} & \boldsymbol{0} & \boldsymbol{0} \end{bmatrix} \begin{bmatrix} \boldsymbol{q}(t) \\ \boldsymbol{f}(t) \\ \boldsymbol{e}_{w}(t) \end{bmatrix}, \quad (35)$$

where I_m is the identity matrix of given dimension.

Using the notations (31), (32) and

$$\boldsymbol{A}^{\bullet} = \begin{bmatrix} \boldsymbol{A} & \boldsymbol{F} & \boldsymbol{0} \\ \boldsymbol{0} & \boldsymbol{0} & \boldsymbol{0} \\ -\boldsymbol{C} & \boldsymbol{0} & \boldsymbol{0} \end{bmatrix}, \ \boldsymbol{B}^{\bullet} = \begin{bmatrix} \boldsymbol{B} \\ \boldsymbol{0} \\ \boldsymbol{0} \end{bmatrix}, \ \boldsymbol{W}^{\bullet} = \begin{bmatrix} \boldsymbol{0} \\ \boldsymbol{0} \\ \boldsymbol{I}_{m} \end{bmatrix}, \ \boldsymbol{C}^{\bullet} = \begin{bmatrix} \boldsymbol{C} & \boldsymbol{0} & \boldsymbol{0} \end{bmatrix}, \quad (36)$$

then the state-space description of the closed-loop system (34), (35), (30) can be written as

$$\dot{\boldsymbol{q}}^{\bullet}(t) = \boldsymbol{A}_{c}^{\bullet} \boldsymbol{q}^{\bullet}(t) + \boldsymbol{W}^{\bullet} \boldsymbol{w}(t), \qquad (37)$$

$$\mathbf{y}(t) = \boldsymbol{C}^{\bullet} \boldsymbol{q}^{\bullet}(t), \qquad (38)$$

where

$$\boldsymbol{A}_{c}^{\bullet} = \boldsymbol{A}^{\bullet} - \boldsymbol{B}^{\bullet} \boldsymbol{K}^{\bullet} \tag{39}$$

is the closed-loop system matrix of the expanded system.

In order to test whether the system is controllable with reference attenuations γ_{∞} , the transfer function matrices

$$\boldsymbol{G}^{\bullet}(s) = \boldsymbol{C}^{\bullet}(s\boldsymbol{I}_{n+p+m} - \boldsymbol{A}_{c}^{\bullet})^{-1}\boldsymbol{B}^{\bullet}, \qquad (40)$$

$$\boldsymbol{G}_{w}(s) = \boldsymbol{C}^{\bullet}(s\boldsymbol{I}_{n+p+m} - \boldsymbol{A}_{c}^{\bullet})^{-1}\boldsymbol{W}^{\bullet}, \qquad (41)$$

are considered in the following.

Lemma 4 (H_2 control synthesis) The state feedback control (30) to the system (37), (38) exists and $\|\mathbf{G}^{\bullet}(s)\|_2 < \gamma_2^{\bullet}$ if there exist symmetric positive definite matrices $\mathbf{V}^{\bullet} \in \mathfrak{R}^{(n+p+m)\times(n+p+m)}$, $\mathbf{H}^{\bullet} \in \mathfrak{R}^{m\times m}$, a matrix $\mathbf{Y}^{\bullet} \in \mathfrak{R}^{r\times(n+p+m)}$ and a positive scalar $\eta^{\bullet} \in \mathfrak{R}$ such that

$$\boldsymbol{V}^{\bullet} = \boldsymbol{V}^{\bullet T} > 0, \qquad \boldsymbol{H}^{\bullet} = \boldsymbol{H}^{\bullet T} > 0, \qquad (42)$$

$$\begin{bmatrix} \mathbf{A}^{\bullet}\mathbf{V}^{\bullet} + \mathbf{V}^{\bullet}\mathbf{A}^{\bullet T} - \mathbf{B}^{\bullet}\mathbf{Y}^{\bullet} - \mathbf{Y}^{\bullet T}\mathbf{B}^{\bullet T} & * \\ \mathbf{B}^{\bullet T} & -\mathbf{I}_{r} \end{bmatrix} < 0, \qquad (43)$$

$$\begin{bmatrix} \mathbf{V}^{\bullet} & * \\ \mathbf{C}^{\bullet}\mathbf{V}^{\bullet} & \mathbf{H}^{\bullet} \end{bmatrix} > 0, \qquad tr(\mathbf{H}^{\bullet}) < \eta^{\bullet}.$$
(44)

When the above conditions hold, the control law gain is

$$\boldsymbol{K}^{\bullet} = \boldsymbol{Y}^{\bullet} (\boldsymbol{V}^{\bullet})^{-1}. \tag{45}$$

Proof Rearranging the inequality (6) by using Schur complement property, it yields

$$\begin{bmatrix} \mathbf{A}\mathbf{V} + \mathbf{V}\mathbf{A}^T & \mathbf{B} \\ \mathbf{B}^T & -\mathbf{I}_r \end{bmatrix} < 0.$$
(46)

Replacing **A** in (46) by (40) and **V** by V^{\bullet} redefines the linear matrix inequality (46) as follows

$$\begin{bmatrix} (\mathbf{A}^{\bullet} - \mathbf{B}^{\bullet} \mathbf{K}^{\bullet}) \mathbf{V}^{\bullet} + \mathbf{V}^{\bullet} (\mathbf{A}^{\bullet} - \mathbf{B}^{\bullet} \mathbf{K}^{\bullet})^{T} & \mathbf{B}^{\bullet} \\ \mathbf{B}^{\bullet T} & -\mathbf{I}_{r} \end{bmatrix} < 0$$
(47)

so, with the notation

$$\boldsymbol{Y}^{\bullet} = \boldsymbol{K}^{\bullet} \boldsymbol{V}^{\bullet}, \tag{48}$$

(47) implies (43).

The objective of H₂ control is to minimize the constraint

$$\operatorname{tr}(\boldsymbol{C}^{\bullet}\boldsymbol{V}^{\bullet}\boldsymbol{C}^{\bullet T}) < (\boldsymbol{\gamma}_{2}^{\bullet})^{2}$$

$$\tag{49}$$

but this inequality cannot be directly optimized. Introducing the inequality

$$\boldsymbol{H}^{\bullet} > \boldsymbol{C}^{\bullet} \boldsymbol{V}^{\bullet} \boldsymbol{C}^{\bullet T} = \boldsymbol{C}^{\bullet} \boldsymbol{V}^{\bullet} (\boldsymbol{V}^{\bullet})^{-1} \boldsymbol{V}^{\bullet} \boldsymbol{C}^{\bullet T}, \qquad \operatorname{tr}(\boldsymbol{H}^{\bullet}) = \boldsymbol{\eta}^{\bullet}, \tag{50}$$

with a new matrix variable H^{\bullet} being symmetric and positive definite, and using Schur complement property, then (49), (50) imply directly (44). This concludes the proof.

Note, to obtain a feasible block structure of the LMI, Schur complement property has to be used. Contrarily to the algorithm presented in [20], the Schur complement property was used to rearrange (6) to obtain (46) while the dual Schur complement property was applied to modify (49). It is the main reason that the proof is attached to this lemma.

Lemma 5 (H_{∞} control synthesis) The state feedback control (30) to the system (37), (38) exists and $\|\boldsymbol{G}_{w}^{\bullet}(s)\|_{\infty} < \gamma_{\infty}$ if there exist a symmetric positive definite matrix $\boldsymbol{Q}^{\bullet} \in \mathfrak{R}^{(n+p+m)\times(n+p+m)}$, a matrix $\boldsymbol{Z}^{\bullet} \in \mathfrak{R}^{r\times(n+p+m)}$ and a positive scalar $\gamma_{\infty}^{\bullet} \in \mathfrak{R}$ such that

$$\boldsymbol{Q}^{\bullet} = \boldsymbol{Q}^{\bullet T} > 0, \qquad (51)$$

$$\begin{bmatrix} \mathbf{A}^{\bullet} \mathbf{Q}^{\bullet} + \mathbf{Q}^{\bullet} \mathbf{A}^{\bullet T} - \mathbf{B}^{\bullet} \mathbf{Z}^{\bullet} - \mathbf{Z}^{\bullet T} \mathbf{B}^{\bullet T} & * & * \\ \mathbf{W}^{\bullet T} & -\gamma_{\infty} \mathbf{I}_{m} & * \\ \mathbf{C}^{\bullet} \mathbf{Q}^{\bullet} & \mathbf{0} & -\gamma_{\infty} \mathbf{I}_{m} \end{bmatrix} < 0.$$
(52)

When the above conditions hold, the control law gain is

$$\boldsymbol{K}^{\bullet} = \boldsymbol{Z}^{\bullet}(\boldsymbol{Q}^{\bullet})^{-1}.$$
(53)

Proof Replacing in (9) \boldsymbol{A} by $\boldsymbol{A}_c^{\bullet}$, \boldsymbol{R} by \boldsymbol{R}^{\bullet} , \boldsymbol{B} by \boldsymbol{W}^{\bullet} and \boldsymbol{I}_r by \boldsymbol{I}_m , and defining the transform matrix

$$T^{\bullet} = \operatorname{diag} \left[\begin{array}{cc} \boldsymbol{Q}^{\bullet} & \boldsymbol{I}_m \end{array} \right], \qquad \boldsymbol{Q}^{\bullet} = (\boldsymbol{R}^{\bullet})^{-1},$$
 (54)

then pre-multiplying the left side and post-multiplying the ride side of (9) by T^{\bullet} , it yields

$$\begin{bmatrix} \mathbf{A}_{c}^{\bullet}\mathbf{Q}^{\bullet} + \mathbf{Q}^{\bullet}\mathbf{A}_{c}^{\bullet T} & * & * \\ \mathbf{W}^{\bullet T} & -\gamma_{\infty}\mathbf{I}_{m} & * \\ \mathbf{C}^{\bullet}\mathbf{Q}^{\bullet} & \mathbf{0} & -\gamma_{\infty}\mathbf{I}_{m} \end{bmatrix} < 0.$$
(55)

Substituting (39) modifies the linear matrix inequality (55) as follows

$$\begin{bmatrix} (\mathbf{A}^{\bullet} - \mathbf{B}^{\bullet}\mathbf{K}^{\bullet})\mathbf{Q}^{\bullet} + \mathbf{Q}^{\bullet}(\mathbf{A}^{\bullet} - \mathbf{B}^{\bullet}\mathbf{K}^{\bullet})^{T} & * & * \\ \mathbf{W}^{\bullet T} & -\gamma_{\infty}\mathbf{I}_{m} & * \\ \mathbf{C}^{\bullet}\mathbf{Q}^{\bullet} & \mathbf{0} & -\gamma_{\infty}\mathbf{I}_{m} \end{bmatrix} < 0$$
(56)

and with the notation

$$\boldsymbol{Z}^{\bullet} = \boldsymbol{K}^{\bullet} \boldsymbol{Q}^{\bullet} \tag{57}$$

 \square

(56) implies (52). This concludes the proof.

It is important to point out, due to the structure of the matrix \mathbf{A}^{\bullet} , \mathbf{B}^{\bullet} that a solution of (51), (52) is usually marginal feasible [15], i.e., $rank(\mathbf{A}^{\bullet} - \mathbf{B}^{\bullet}\mathbf{K}^{\bullet}) < n + p + m$, and the design has to be combined with any constraint. For this reason the proof for this lemma is given. The mixed H₂/H_{∞} control principle, using H₂ and H_{∞} performance constraints to solve FTC problem, is proposed in [27].

5. Joint design of fault tolerant control

With these expressions it is now easy to formulate the joint approach for integrated design of fault estimation and FTC, where $q^{\bullet T}(t)$ is considered as

$$\boldsymbol{q}^{\bullet T}(t) = \begin{bmatrix} \boldsymbol{q}^{T}(t) & \boldsymbol{f}_{e}^{T}(t) & \boldsymbol{e}_{w}^{T}(t) \end{bmatrix}.$$
(58)

Theorem 1 The state feedback control (30), (58) to the system (38), (39) exists and $\|\boldsymbol{G}^{\bullet}(s)\|_{2} < \gamma_{2}$ as well as $\|\boldsymbol{G}^{\bullet}_{w}(s)\|_{\infty} < \gamma_{\infty}$ if for given positive scalars $a, \rho \in \mathfrak{R}, a > \rho > 0$, there exist symmetric positive definite matrices $\boldsymbol{V}^{\bullet} \in \mathfrak{R}^{(n+p+m)\times(n+p+m)}$, $\boldsymbol{P}^{\bullet} \in \mathfrak{R}^{(n+p)\times(n+p)}$, $\boldsymbol{H}^{\bullet} \in \mathfrak{R}^{m\times m}$, matrices $\boldsymbol{Y}^{\bullet} \in \mathfrak{R}^{r\times(n+p+m)}$, $\boldsymbol{S}^{\circ} \in \mathfrak{R}^{(n+p)\times m}$ and a positive scalar $\gamma^{\bullet}_{\infty} \in \mathfrak{R}$ such that

$$\boldsymbol{V}^{\bullet} = \boldsymbol{V}^{\bullet T} > 0, \qquad \boldsymbol{P}^{\bullet} = \boldsymbol{P}^{\bullet T} > 0, \qquad \boldsymbol{H}^{\bullet} = \boldsymbol{H}^{\bullet T} > 0, \tag{59}$$

$$\begin{bmatrix} -\rho \boldsymbol{P}^{\circ} & \ast \\ \boldsymbol{P}^{\circ} \boldsymbol{A}^{\circ} - \boldsymbol{S}^{\circ} \boldsymbol{C}^{\circ} + a \boldsymbol{P}^{\circ} & -\rho \boldsymbol{P}^{\circ} \end{bmatrix} < 0,$$
(60)

$$\begin{bmatrix} \mathbf{A}^{\bullet}\mathbf{V}^{\bullet} + \mathbf{V}^{\bullet}\mathbf{A}^{\bullet T} - \mathbf{B}^{\bullet}\mathbf{Y}^{\bullet} - \mathbf{Y}^{\bullet T}\mathbf{B}^{\bullet T} & * & * \\ \mathbf{W}^{\bullet T} & -\gamma_{\infty}\mathbf{I}_{m} & * \\ \mathbf{C}^{\bullet}\mathbf{V}^{\bullet} & \mathbf{0} & -\gamma_{\infty}\mathbf{I}_{m} \end{bmatrix} < 0, \quad (61)$$

$$\begin{bmatrix} \mathbf{A}^{\bullet}\mathbf{V}^{\bullet} + \mathbf{V}^{\bullet}\mathbf{A}^{\bullet T} - \mathbf{B}^{\bullet}\mathbf{Y}^{\bullet} - \mathbf{Y}^{\bullet T}\mathbf{B}^{\bullet T} & * \\ \mathbf{B}^{\bullet T} & -\mathbf{I}_{r} \end{bmatrix} < 0,$$
(62)

$$\begin{bmatrix} \mathbf{V}^{\bullet} & \ast \\ \mathbf{C}^{\bullet}\mathbf{V}^{\bullet} & \mathbf{H}^{\bullet} \end{bmatrix} > 0, \qquad tr(\mathbf{H}^{\bullet}) < \eta^{\bullet}.$$
(63)

When the above conditions hold

$$\boldsymbol{K}^{\bullet} = \boldsymbol{Y}^{\bullet} (\boldsymbol{V}^{\bullet})^{-1}, \qquad \boldsymbol{J}^{\circ} = (\boldsymbol{P}^{\circ})^{-1} \boldsymbol{S}^{\circ}.$$
(64)

Proof Replacing in (10) \boldsymbol{A} by $\boldsymbol{A}_{e}^{\circ}$ and \boldsymbol{P} by \boldsymbol{P}° results

$$\begin{bmatrix} -\rho \boldsymbol{P}^{\circ} & \ast \\ \boldsymbol{P}^{\circ} (\boldsymbol{A}^{\circ} - \boldsymbol{J}^{\circ} \boldsymbol{C}^{\circ}) + a \boldsymbol{P}^{\circ} & -\rho \boldsymbol{P}^{\circ} \end{bmatrix} < 0$$
(65)

and with the notation

$$\mathbf{S}^{\circ} = \mathbf{P}^{\circ} \mathbf{J}^{\circ} \tag{66}$$

(65) implies (60).

Prescribing a unique solution of K^{\bullet} with respect to (45) and (53) that is

$$\boldsymbol{V}^{\bullet} = \boldsymbol{Q}^{\bullet}, \qquad \boldsymbol{Y}^{\bullet} = \boldsymbol{Z}^{\bullet}, \tag{67}$$

then (42)-(44) and (51), (52) in the joint sense implies (61)-(64). This concludes the proof. $\hfill \Box$

Note, the introduced mixed H_2/H_{∞} control maximizes the H_2 norm over all statefeedback gains K^{\bullet} while the H_{∞} norm constraint is minimized. Comparing with [20], the set of LMIs (61)-(63) is well conditioned and feasible. The main reason for the use of D-stability principle in the fault observer design is to adapt interactive the fault observer dynamics to the dynamics of the FTC structure.

6. Illustrative example

To illustrate the proposed method, a system whose dynamics is described by equations (1), (2) is considered with the matrix parameters [12]

$$\mathbf{A} = \begin{vmatrix} 1.380 & -0.208 & 6.715 & -5.676 \\ -0.581 & -4.290 & 0.000 & 0.675 \\ 1.067 & 4.273 & -6.654 & 5.893 \\ 0.048 & 4.273 & 1.343 & -2.104 \end{vmatrix}$$

$$\boldsymbol{B} = \begin{bmatrix} 0.000 & 0.000 \\ 5.679 & 0.000 \\ 1.136 & -3.146 \\ 1.136 & 0.000 \end{bmatrix}, \ \boldsymbol{F} = \begin{bmatrix} 1.400 \\ 1.504 \\ 2.233 \\ 0.610 \end{bmatrix}, \ \boldsymbol{C} = \begin{bmatrix} 4 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \end{bmatrix}.$$

Solving (59)–(63) using SeDuMi package, the design problem is solved as feasible where, with the prescribed stability region parameters a = 10, $\rho = 8.5$, the resulted control system parameters are

$$\boldsymbol{K}_{q} = \begin{bmatrix} 0.0709 & -0.7022 & 0.0245 & 1.9807 \\ -14.6220 & -0.0706 & -1.2190 & -1.2406 \end{bmatrix},$$
$$\boldsymbol{K}_{f} = \begin{bmatrix} 0.2546 \\ -0.5672 \end{bmatrix}, \quad \boldsymbol{K}_{w} = \begin{bmatrix} -0.0859 & -0.3923 \\ 0.9172 & 0.0261 \end{bmatrix},$$
$$\boldsymbol{J} = \begin{bmatrix} 3.3817 & -2.7407 \\ 0.4280 & 2.6900 \\ 1.2510 & 7.4649 \\ 0.5128 & 8.2180 \end{bmatrix}, \quad \boldsymbol{L} = \begin{bmatrix} 1.8415 & 3.1383 \end{bmatrix}, \quad \boldsymbol{H} = \begin{bmatrix} 31.3538 & 1.1875 \\ 1.1875 & 21.0385 \end{bmatrix},$$
$$\boldsymbol{\gamma}_{\infty} \ge 21.1631, \qquad \boldsymbol{\gamma}_{2}^{2} \ge 28.0805,$$

while

$$\begin{split} \rho(\boldsymbol{A}_{c}^{\circ}) &= \{0, \ -0.1784, \ -0.3017 \ -1.6245 \pm 7.0670 \, i, \ -5.0321 \pm 16.4998 \, i\} \,, \\ \rho(\boldsymbol{A}_{e}^{\circ}) &= \{-3.4866 \ -5.9462 \pm 1.1756 \, i, \ -9.6424 \pm 0.6226 \, i\} \,, \end{split}$$

which affirms that for given (a, ρ) the dynamics of the fault augmented observer is substantially faster than the dynamics of the FTC structure.

The faults in simulations are generated using the window

$$g(t) = \begin{cases} 0, & t \leq t_{sa}, \\ \frac{1}{t_{sb} - t_{sa}}(t - t_{sa}), & t_{sa} < t_{sb}, \\ 1, & t_{sb} \leq t_{ea}, \\ -\frac{1}{t_{eb} - t_{ea}}(t - t_{eb}), & t_{ea} < t_{eb}, \\ 0, & t \geq t_{eb}, \end{cases}$$

where it is adjusted

$$f_1(t) = g(t)\sin(\omega t), \quad \omega = 0.2 \text{ rad/sec}, \quad f_2(t) = g(t),$$

 $t_{sa} = 30s, \quad t_{sb} = 35s, \quad t_{ea} = 65s, \quad t_{eb} = 70s,$

and the initial conditions are

$$\boldsymbol{q}^{\bullet}(0) = \boldsymbol{0}, \ \boldsymbol{q}^{\circ}(0) = \boldsymbol{0}, \ \boldsymbol{w}^{T}(t) = \begin{bmatrix} 1 & 2 \end{bmatrix}.$$

Figures 2-4 show the sinusoidal-like fault and its estimate, the corresponding estimated closed-loop system outputs as well as the corresponding closed-loop system outputs with fault compensation, when the sinusoidal-like fault arises. In the same manner, Figures 1-6 characterize the signals in the FTC structure with integrated fault estimation when the ramp fault comes into.

From the above given figures it is obvious that the integrated FTC, which parameters are obtained as a solution of the LMI problem specified by Theorem 1, can with sufficient precision approximate given class of slowly warring faults that their impact on the system output variables is successfully compensated. Moreover, the mixed joint design, which includes the updated \mathcal{D} -stability circle criterion, outperforms the two-stage design approach without increasing conservatism.

7. Concluding remarks

A modified approach for designing fault augmented observes, integrated with the compensation FTC structure, is presented in the contribution. Using LMI technique, the exploited mixed H_2/H_{∞} control design to regularize in general marginally feasible conditions, a dual form of the Schur complement to linearize the bilinear γ_2 constraint in LMI approach for H_2 control principle, as well as an updated \mathcal{D} -stability circle criterion in fault observer design to adapt the fault observer dynamics to the dynamics of the control system, the design conditions are established as feasible problem, accomplishing under given quadratic constraints. Presented illustrative example confirms the effectiveness of the proposed design alternative, to construct the control structure with sufficient approximation of given class slowly warring faults and compensation of their impact on the system output variables.







Figure 2: Estimated sinusoidal-like fault







Figure 4: Closed-loop system outputs with fault compensation



Figure 5: Estimated closed-loop system outputs



Figure 6: Closed-loop system outputs with fault compensation

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